9th Triennial International Aircraft Fire and Cabin Safety Research Conference October 28 – October 31, 2019 Atlantic City, NJ.

Comparison of Dynamic Responses of 50th percentile Hybrid II and FAA Hybrid III Anthropometric Test Devices (ATD) during Aircraft Seat Tests

Authors:

Name: Prasanna Bhonge

Affiliation: Gulfstream Aerospace Co.

Name: Richard DeWeese

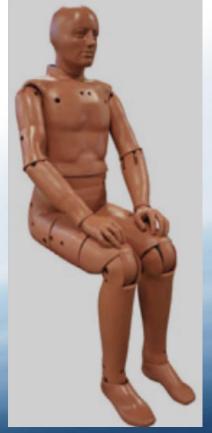
Affiliation: FAA Civil Aerospace Medical Institute, Engineering Sciences Section

Name: David Moorcroft

Affiliation: FAA Civil Aerospace Medical Institute, Biodynamics Research Team

Name: Raki Islam

Affiliation: Safran Seats



Hybrid II



FAA Hybrid III

Background

History:

- Aircraft seat dynamic qualification tests require use of a Hybrid II ATD or equivalent.
- Auto crash tests now use the Hybrid III ATD, which is a more advanced and biofidelic test dummy.
- A version of the Hybrid III was developed that has been approved by the FAA as equivalent to the Hybrid II [Ref 1].
- Although this ATD (called the FAA Hybrid III) has been available since 2000 it has not been widely used by the aviation industry.
- This lack of acceptance may be due to Industry's concerns that the ATD could interact with the seat/surroundings differently than with the Hybrid II and produce significantly different results.

Purpose:

 The purpose of this project was to review available FAA and Industry data to evaluate the actual differences between Hybrid II and FAA Hybrid III ATD response during dynamic seat tests.

Agenda

ATD Comparison

Geometry Weight

ATD Dynamic Performance

Repeatability

Rigid Seat Vertical Test

Rigid Seat Horizontal Test

Real Seat Vertical Test

Real Seat Horizontal Test

Test results comparison

Rigid Seat Vertical Test

Rigid Seat Horizontal Test

Real Seat Vertical Test

Real Seat Horizontal Test

Findings

Conclusions

Comparison of Standard 50th Percentile ATDs

Hybrid II (1972)

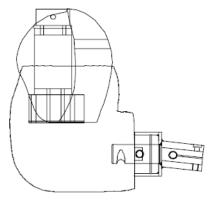


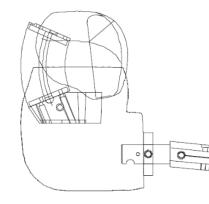
Hybrid III (1986)

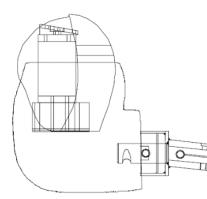


FAA Hybrid III (2000)









FAA Hybrid III ATD is predominantly made up of Hybrid III ATD parts except:

- Hybrid II lumbar spine
- Hybrid II abdominal insert
- Hybrid II chest jacket
- Hybrid II upper leg bone
- Hybrid II lumbar load cell and pelvic adaptor block
- Custom thorax/lumbar adaptor

Reference: [1]

ATD Mass and Dimension Comparison

| BODY COMPONENT/ SEGMENT | HYBRID II | FAA HYBRID III |
|--|----------------|-------------------|
| HEAD | 11.2 | 10 |
| UPPER TORSO (INCLUDING LUMBAR SPINE) | 41.5 | 41.3 |
| LOWER TORSO (INCLUDING VISCERAL SAC AND UPPER THIGHS) | 37.5 | 37.9 |
| UPPER ARM (BOTH) | 9.6 | 8.8 |
| LOWER ARM (BOTH) | 6.8 | 7.5 |
| HAND (BOTH) | 2.8 | 2.5 |
| UPPER LEG (BOTH) | 35.2 | 34 |
| LOWER LEG AND FOOT (BOTH) | 19.4 | 24 |
| TOTAL (INCLUDING INSTRUMENTATION IN HEAD, CHEST, AND FEMURS) | 164 (lb) | 166 (lb) |
| SPECIFIED WEIGHT | 164 ±3 (lb) | 164 ±3 (LB) |
| SPECIFIED SITTING HEIGHT | 35.7 ±0.1 (in) | 35.7 ±0.1 (in) |
| HIP POINT HEIGHT TOLERANCE | 0.4 (in) | 0.14 (in) |

- Specified total weight and sitting height are identical
- Segment weights are similar
- The pelvis manufacturing tolerances that control the thickness of rubber/foam on the bottom of the pelvis are different

Reference: [1, 4, 5]

ATD Dynamic Performance

ATD Repeatability

Rigid Seat Vertical Test Rigid Seat Horizontal Test Real Seat Vertical Test Real Seat Horizontal Test

ATD-to-ATD Comparison

Rigid Seat Vertical Test Rigid Seat Horizontal Test Real Seat Vertical Test Real Seat Horizontal Test

Vertical Test – Rigid Seat, Thin Cushion – Hybrid II

Test repeatability: Lumbar load

| ATD | Test Number | Goal G | Peak G | Peak Lumbar (lb) | Normalized Lumbar (lb) | Test Repeatability |
|-----------|----------------|-----------|--------|------------------------|---------------------------|-----------------------|
| Hybrid II | A12013 | 9 | 9.9 | 635 | 580 | 4.7% |
| Hybrid II | A12031 | 9 | 10.2 | 630 | 553 | 4.7 /0 |
| | | | | | | |
| Hybrid II | A12011 | 14 | 14.5 | 940 | 909 | 12 60/ |
| Hybrid II | A12032 | 14 | 15.5 | 1148 | 1040 | 12.6% |
| | | | | | | |
| Hybrid II | A12012 | 19 | 20.0 | 1954 | 1860 | |
| Hybrid II | A12014 | 19 | 19.4 | 1866 | 1827 | 8% |
| Hybrid II | A12033* | 19 | 18.4 | 1919 | 1986 | |
| *Pulse Fa | ailed | | | | | |

Reference: [2]

| ATD | Test Number | Goal G | Peak G | Peak Lumbar (lb) | Normalized Lumbar (14G) (lb) | Test Repeatability |
|-----------|----------------|-----------|--------|------------------------|------------------------------------|-----------------------|
| Hybrid II | 96041 | 14 | 16.0 | 1362 | 1195 | |
| Hybrid II | 96042 | 14 | 16.0 | 1355 | 1186 | 3.3% |
| Hybrid II | 96043 | 14 | 15.6 | 1288 | 1155 | |

Test Repeatability: 3.3% – 12.6%



Test Repeatability = $\frac{\text{Maximum Load} - \text{Minimum Load}}{\text{Maximum Load}} * 100\%$

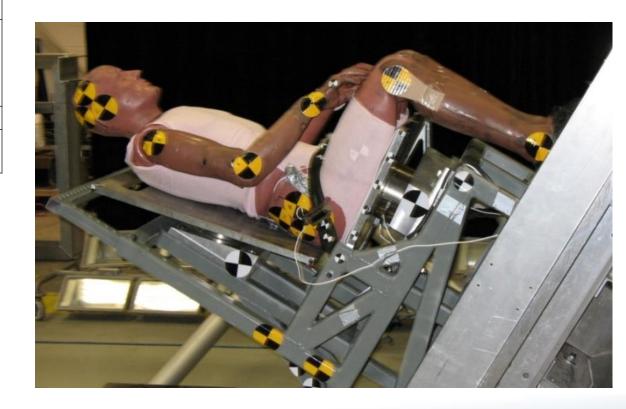
Reference: [3]

Vertical Test – Rigid Seat, No Cushion – Hybrid II

Test repeatability: Lumbar load

| ATD | Teflon (# of sheets) | Test Number | Goal G | Peak G | Peak Lumbar (lb) | Normalized Lumbar (lb) | Test Repeatability |
|-----------|----------------------------|----------------|-----------|--------|------------------------|------------------------------|-----------------------|
| Hybrid II | 0 | 06165-5 | 14 | 14.7 | 858 | 817 | |
| Hybrid II | 0 | 06165-6 | 14 | 14.6 | 960 | 921 | 13.0% |
| Hybrid II | 0 | 06165-25 | 14 | 14.65 | 837 | 800 | 13.076 |
| Hybrid II | 0 | 06165-26 | 14 | 14.35 | 935 | 912 | |
| | | | | | | | |
| Hybrid II | 2 | 07324-11 | 19 | 19.66 | 1757 | 1698 | 2.5% |
| Hybrid II | 2 | 07324-12 | 19 | 19.43 | 1693 | 1655 | 2.5% |

Test Repeatability: 2.5% – 13.0%



Test Repeatability = $\frac{\text{Maximum Load-Minimum Load}}{\text{Maximum Load}} * 100\%$

Reference: [3,4,5]

Vertical Test – Rigid Seat, w & w/o Cushion – FAA Hybrid III

Test repeatability: Lumbar load

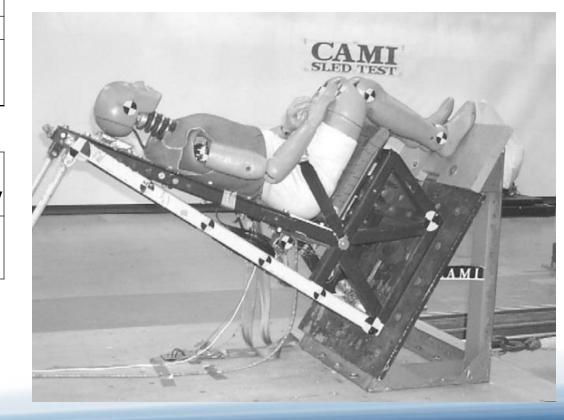
| ATD | Teflon (# of sheets) | Test Number | Goal G | Peak G | Peak Lumbar (lb) | Normalized Lumbar (lb) | Test Repeatability |
|----------------|----------------------------|----------------|-----------|-----------|------------------------|---------------------------|-----------------------|
| FAA Hybrid III | 2 | 07324-13 | 19 | 19.08 | 1713 | 1705 | |
| FAA Hybrid III | 2 | 07324-14 | 19 | 19.14 | 1736 | 1723 | 4.3 |
| FAA Hybrid III | 2 | 07324-15 | 19 | 19.18 | 1798 | 1781 | |
| | | | | | | | |
| FAA Hybrid III | NA | 06165-7 | 14 | 14.6 | 1013 | 971 | |
| FAA Hybrid III | NA | 06165-8 | 14 | 14.8 | 1028 | 972 | 6.7 |
| FAA Hybrid III | NA | 06165-28 | 14 | 14.28 | 924 | 906 | |

| ATD | Teflon (# of sheets) | Test Number | Goal G | Peak G | Peak Lumbar (lb) | Normalized Lumbar (lb) | Test Repeatability |
|----------------|----------------------------|----------------|--------|--------|------------------------|---------------------------|-----------------------|
| FAA Hybrid III | NA | 98032 | 14 | 15.0 | 1236 | 1154 | |
| FAA Hybrid III | NA | 98033 | 14 | 15.2 | 1275 | 1174 | 4.3 |
| FAA Hybrid III | NA | 99010 | 14 | 14.8 | 1275 | 1206 | |

Test Repeatability = $\frac{\text{Maximum Load-Minimum Load}}{\text{Maximum Load}} * 100\%$

Reference: [3,4,5]

Test Repeatability: 4.3% – 6.7%



Lumbar Load Comparison - Rigid Seat, Thin Cushion

Hybrid II vs FAA Hybrid III

| ATD | Test Number | Goal G | Peak G | Peak Lumbar (lb) | Normalized Lumbar (lb) | Average | Relative Error |
|----------------|----------------|-----------|--------|---------------------|---------------------------|---------|-------------------|
| Hybrid II | A12013 | 9 | 9.9 | 635 | 580 | 567 | |
| Hybrid II | A12031 | 9 | 10.2 | 630 | 553 | 307 | -8.3% |
| FAA Hybrid III | A12028 | 9 | 9.9 | 573 | 520 | 520 | |
| Hybrid II | A12011 | 14 | 14.5 | 940 | 909 | 975 | |
| Hybrid II | A12032 | 14 | 15.5 | 1148 | 1040 | 975 | -10.4% |
| FAA Hybrid III | A12029 | 14 | 15.0 | 939 | 874 | 874 | |
| Hybrid II | A12012 | 19 | 20.0 | 1954 | 1860 | | |
| Hybrid II | A12014 | 19 | 19.4 | 1866 | 1827 | 1891 | -4.5% |
| Hybrid II | A12033* | 19 | 18.4 | 1919 | 1986 | | -4.5% |
| FAA Hybrid III | A12030* | 19 | 18.7 | 1774 | 1806 | 1806 | |
| *Pulse Failed | | • | | | | | |

Reference: [2]

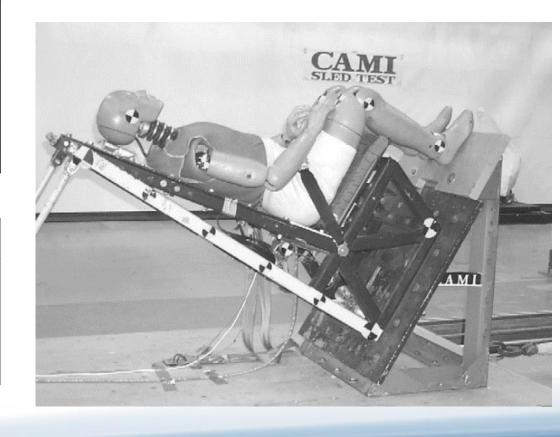
| ATD | Test Number | Goal G | Peak G | Peak Lumbar (lb) | Normalized Lumbar (lb) | Average | Relative Error |
|----------------|----------------|-----------|--------|---------------------|---------------------------|---------|-------------------|
| Hybrid II | 96041 | 14 | 16.0 | 1362 | 1195 | | |
| Hybrid II | 96042 | 14 | 16.0 | 1355 | 1186 | 1179 | |
| Hybrid II | 96043 | 14 | 15.6 | 1288 | 1155 | | 0.0% |
| FAA Hybrid III | 98032 | 14 | 15.0 | 1236 | 1154 | | 0.070 |
| FAA Hybrid III | 98033 | 14 | 15.2 | 1275 | 1174 | 1178 | |
| FAA Hybrid III | 99010 | 14 | 14.8 | 1275 | 1206 | | |

Relative Error

Avg(FAA Hybrid III Lumbar Load) — Avg(Hybrid II Lumbar Load)

Reference: [3] = Avg(Hybrid II Lumbar Load)

Relative Error: 0.0% – -10.4%

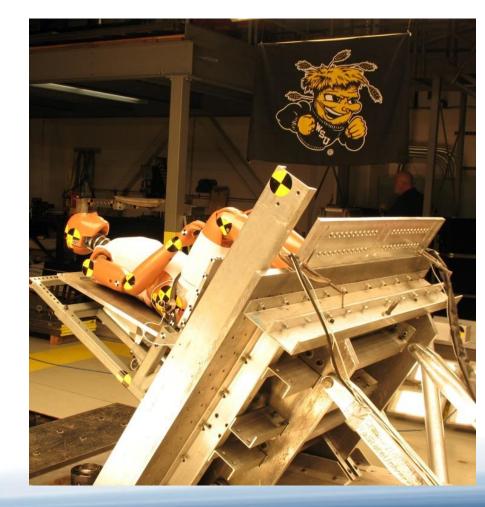


Lumbar Load Comparison - Rigid Seat, No Cushion

Hybrid II vs FAA Hybrid III

Relative Error: 3.6% – 10.2%

| ATD | Teflon (# of sheets) | Test Number | Goal G | Peak G | Peak Lumbar (lb) | Normalized Lumbar (lb) | Average | Relative Error |
|----------------|----------------------|----------------|-----------|-----------|------------------------|------------------------------|---------|-------------------|
| Hybrid II | 2 | 07324-11 | 19 | 19.66 | 1757 | 1698 | 1677 | |
| Hybrid II | 2 | 07324-12 | 19 | 19.43 | 1693 | 1655 | 1077 | |
| | | | | | | | | 3.6% |
| FAA Hybrid III | 2 | 07324-13 | 19 | 19.08 | 1713 | 1705 | | J.0 /0 |
| FAA Hybrid III | 2 | 07324-14 | 19 | 19.14 | 1736 | 1723 | 1737 | |
| FAA Hybrid III | 2 | 07324-15 | 19 | 19.18 | 1798 | 1781 | | |
| | | | | | | | | |
| | | | | | | | | |
| Hybrid II | 0 | 06165-5 | 14 | 14.7 | 858 | 817 | | |
| Hybrid II | 0 | 06165-6 | 14 | 14.6 | 960 | 921 | 862 | |
| Hybrid II | 0 | 06165-25 | 14 | 14.65 | 837 | 800 | 002 | |
| Hybrid II | 0 | 06165-26 | 14 | 14.35 | 935 | 912 | | 10.2% |
| | | | | | | | | 10.27 |
| FAA Hybrid III | 0 | 06165-7 | 14 | 14.6 | 1013 | 971 | | |
| FAA Hybrid III | 0 | 06165-8 | 14 | 14.8 | 1028 | 972 | 950 | |
| FAA Hybrid III | 0 | 06165-28 | 14 | 14.28 | 924 | 906 | | |



 $Relative Error = \frac{Avg(FAA \ Hybrid \ III \ Lumbar \ Load) - Avg(Hybrid \ II \ Lumbar \ Load)}{Avg(Hybrid \ II \ Lumbar \ Load)}$

Reference: [3, 4]

Lumbar Load Repeatability - Real Seat

Hybrid II vs FAA Hybrid III

| ATD | Test Number | Goal G | Peak G | Peak Lumbar (lb) | Normalized Lumbar (lb) | Test Repeatability |
|----------------|----------------|-----------|-----------|------------------------|---------------------------|-----------------------|
| Hybrid II | A99017 | 14 | 14.2 | | 1617 | |
| Hybrid II | A99018 | 14 | 14.4 | | 1503 | 7.1% |
| | | | | | | |
| FAA Hybrid III | A99014 | 14 | 13.6 | | 1590 | |
| FAA Hybrid III | A99015 | 14 | 14.2 | | 1622 | 2.0% |

Test Repeatability: 2.0% – 7.1%



Test Repeatability = $\frac{\text{Maximum Load-Minimum Load}}{\text{Maximum Load}} * 100\%$

Reference: [9]

Lumbar Load Comparison - Real Seat

Hybrid II vs FAA Hybrid III

| ATD | Test Number | Goal G | Peak G | Peak Lumbar (lb) | Normalized Lumbar (lb) | Average | Relative Error |
|----------------|----------------|-----------|-----------|---------------------|---------------------------|---------|-------------------|
| Hybrid II | A99017 | 14 | 14.2 | | 1617 | | |
| Hybrid II | A99018 | 14 | 14.4 | | 1503 | 1560 | |
| | | | | | | | 3.0% |
| FAA Hybrid III | A99014 | 14 | 13.6 | | 1590 | | |
| FAA Hybrid III | A99015 | 14 | 14.2 | | 1622 | 1606 | |

Reference: [9]

| ATD | Test Number | Goal G | Peak G | Peak Lumbar (lb) | Normalized Lumbar (lb) | Average | Relative Error |
|----------------|----------------|-----------|-----------|---------------------|---------------------------|---------|-------------------|
| Hybrid II | | 14 | | | 1406 | 1406 | |
| FAA Hybrid III | | 14 | | | 1437 | 1437 | 2.2% |

Reference: [8]

 $Relative Error = \frac{Avg(FAA \ Hybrid \ III \ Lumbar \ Load) - Avg(Hybrid \ II \ Lumbar \ Load)}{Avg(Hybrid \ II \ Lumbar \ Load)}$

Relative Error: 2.2% – 3.0%



Lumbar Load Comparison - Oblique Bus. Class

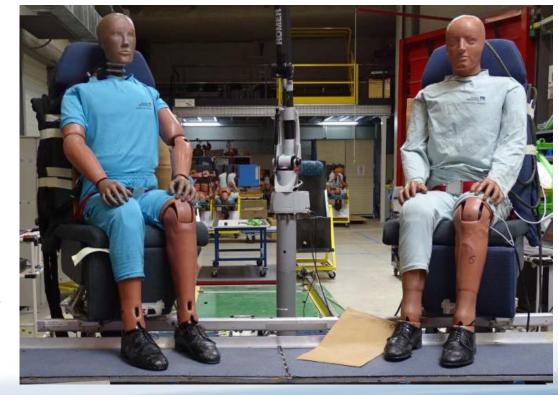
Hybrid II vs FAA Hybrid III

| ATD | Test Number | Goal G | Peak G | Peak Lumbar (lb) | Normalized Lumbar (lb) | Average | Relative Error |
|----------------|----------------|-----------|--------|---------------------|---------------------------|---------|-------------------|
| Hybrid II | | 14 | | | 1202 | 4200 E | |
| Hybrid II | | 14 | | | 1217 | 1209.5 | |
| | | | | | | | 31.4% |
| FAA Hybrid III | | 14 | | | 1551 | 4500 F | |
| FAA Hybrid III | | 14 | | | 1628 | 1589.5 | |

 $\label{eq:Relative Error} \begin{aligned} \text{Relative Error} &= \frac{\text{Avg}(\text{FAA Hybrid III Lumbar Load}) - \text{Avg}(\text{Hybrid II Lumbar Load})}{\text{Avg}(\text{Hybrid II Lumbar Load})} \end{aligned}$

Reference: [8]

Relative Error: 31.4%



Findings: Lumbar Load

Rigid Seat

Test Repeatability:

Hybrid II = 3% to 13%FAA Hybrid III = 4% to 7%

ATD Comparison Relative Error:

1996 Test Series: 0%

2006 Test Series: +4% to +10%

2012 Test Series: -4% to -10%

Overall Repeatability is 13%

Overall Error is ± 10%

Real seats*

Test Repeatability:

Hybrid II = 7%FAA Hybrid III = 2%

Comparison Relative Error:

1999 Test Series: +3%

2019 Test Series: +2%

2019 Oblique Test Series: +31%

^{*} based on data from a very limited number of tests

Horizontal Test – Rigid Seat, Thin Cushion – Hybrid II

Hybrid II repeatability: Head path/ Belt load

Test Repeatability:

Head path: 0 - 0.4 inch

Lap belt: 0.5% – 3.0%

Torso strap load: 4.8%

| Test Config. | ATD | Test Number | Goal G | Peak G | Head Excursion (in) | Test repeatability (in) | Lap Belt Load (lb) | Test repeatability | Torso Strap Load (lb) | Test repeatability |
|-------------------------|---------------|----------------|-----------|-----------|---------------------------|-------------------------|-----------------------|--------------------|--------------------------|--------------------|
| Lap Belt | 1 1- 4: -1 11 | A99001 | 18 | 18.1 | 39.9 | 0.4 | 2100 | 0.00/ | N/A | N1/A |
| Horizontal | Hybrid II | A99002 | 18 | 17.9 | 40.3 | 0.4 | 2164 | 3.0% | N/A | - N/A |
| 4- Point | | A99003 | 16 | 16.3 | 19.5 | | 1839 | a =a/ | 1050 | 4.00/ |
| Restraint Horizontal | - | A99004 | 16 | 16.1 | 19.5 | 0.0 | 1830 | 0.5% | 1000 | 4.8% |

Comment: Although the sled peak acceleration varied somewhat between the tests, the data was not normalized.

$$Test \ Repeatability = \frac{Maximum \ Load - Minimum \ Load}{Maximum \ Load} * 100\%$$

Reference: [1]



Horizontal Test - Rigid Seat, No Cushion - Hybrid II

Hybrid II repeatability: Head path/ Belt load

| ATD | Test Number | Belt Type | Goal G | Head Excursion (in) | Left Lap Belt Force (lb) | Right Lap Belt Force (lb) | Shoulder Belt Force - Left (lb) | Shoulder Belt Force - Right (lb) |
|-----------|----------------|--------------|-----------|---------------------------|--------------------------------|---------------------------------|---------------------------------------|--|
| Hybrid II | 07324-4 | 2 | 16 | 35.2 | 1887.0 | 1870.0 | | |
| Hybrid II | 07324-7 | 2 | 16 | 35.7 | 2002.0 | 1952.0 | | |
| Hybrid II | 07324-28 | 2 | 16 | 36.2 | 1989.0 | 1976.0 | | |
| | | | | | | | | |
| Hybrid II | 07324-8 | 3 | 21 | 18.9 | 1724.0 | 2475.0 | 1606.0 | |
| Hybrid II | 07324-9 | 3 | 21 | 18.9 | 1720.0 | 2509.0 | 1641.0 | |
| Hybrid II | 07324-29 | 3 | 21 | 19.1 | 1599.0 | 2474.0 | 1574.0 | |
| | | | | | | | | |
| Hybrid II | 07324-5 | 4 | 21 | 14.7 | 1904.0 | 1858.0 | 990.0 | 935.0 |
| Hybrid II | 07324-26 | 4 | 21 | 13.4 | 2024.0 | 1958.0 | 920.0 | 995.0 |
| Hybrid II | 07324-27 | 4 | 21 | 13.5 | 2042.0 | 1941.0 | 874.0 | 973.0 |

| ATD | Belt Type | Goal G | Head Excursion (in) | Left Lap Belt Force | Right Lap Belt Force | Shoulder Belt Force – Left | Shoulder Belt Force - Right |
|-----------|--------------|-----------|---------------------------|------------------------|-------------------------|----------------------------------|-----------------------------------|
| Hybrid II | 2 | 16 | 1.0 | 5.7% | 5.4% | N/A | N/A |
| | | | | | | | |
| Hybrid II | 3 | 21 | 0.2 | 7.3% | 1.4% | 4.1% | N/A |
| | | | | | | | |
| Hybrid II | 4 | 21 | 1.3 | 6.7% | 5.1% | 11.7% | 6.0% |

Comment: Although the sled peak acceleration varied somewhat between the tests, the data was not normalized.

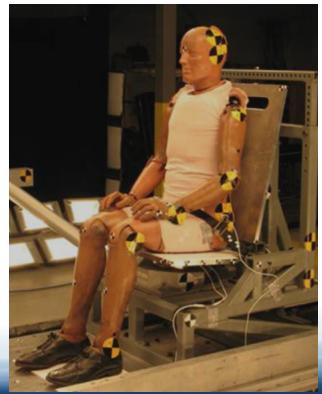
Reference: [4] Test Repeatability = $\frac{\text{Maximum Load} - \text{Minimum Load}}{\text{Maximum Load}} * 100\%$

Test Repeatability:

Head path: 0.2 – 1.3 inch

Lap belt: 1.4% – 7.3%

Torso strap load: 4.1% – 11.7%



Horizontal Test – Rigid Seat, Thin Cushion – FAA Hybrid III

FAA Hybrid III repeatability: Head path/ Belt load

| Test Config. | ATD | Test Number | Goal G | Peak G | Head Excursion (in) | Lap Belt Load (lb) | Torso Strap Load (lb) |
|----------------------|------------|----------------|-----------|-----------|---------------------------|-----------------------|--------------------------|
| Lap Belt | FAA Hybrid | A98045 | 18 | 18.1 | 40.6 | 2174 | NA |
| Horizontal | III | A98046 | 18 | 17.9 | 40.6 | 2208 | NA |
| 4- Point | FAA Hybrid | A99005 | 16 | 16.3 | 19.2 | 1984 | 900 |
| Restraint Horizontal | III | A99006 | 16 | 15.9 | 19.5 | 2068 | 950 |

| ATD | Belt Type | Goal G | Head Excursion (in) | Lap Belt Force | Shoulder Belt Force - Right |
|----------------|--------------|--------|---------------------------|-------------------|-----------------------------------|
| FAA Hybrid III | 2 | 18 | 0.0 | 1.5% | N/A |
| | | | | | |
| FAA Hybrid III | 4 | 16 | 0.3 | 4.1% | 5.3% |

Comment: Although the sled peak acceleration varied somewhat between the tests, the data was not normalized.

Test Repeatability:

Head path: 0.0 - 0.3 inch

Lap belt: 1.5% – 4.1%

Torso strap load: 5.3%



Horizontal Test - Rigid Seat, No Cushion - FAA Hybrid III

FAA Hybrid III repeatability: Head path/ Belt load

| ATD | Test Number | Belt Type | Goal G | Head Excursion (in) | Left Lap Belt Force (lb) | Right Lap Belt Force (lb) | Shoulder Belt Force - Left (lb) | Shoulder Belt Force - Right (lb) |
|----------------|----------------|--------------|-----------|---------------------------|--------------------------------|---------------------------------|---------------------------------------|--|
| FAA Hybrid III | 07324-16 | 2 | 16 | 36.9 | 1851.0 | 1776.0 | | |
| FAA Hybrid III | 07324-17 | 2 | 16 | 37.1 | 1837.0 | 1812.0 | | |
| FAA Hybrid III | 07324-18 | 2 | 16 | 37.4 | 1817.0 | 1783.0 | | |
| | | | | | | | | |
| FAA Hybrid III | 07324-20 | 3 | 21 | 19.5 | 1718.0 | 2324.0 | 1712.0 | |
| FAA Hybrid III | 07324-24 | 3 | 21 | 21.1 | 1713.0 | 2420.0 | 1586.0 | |
| FAA Hybrid III | 07324-25 | 3 | 21 | 20.5 | 1727.0 | 2471.0 | 1637.0 | |
| | | | | | | | | |
| FAA Hybrid III | 07324-21 | 4 | 21 | 16.0 | 1865.0 | 1870.0 | 854.0 | 790.0 |
| FAA Hybrid III | 07324-22 | 4 | 21 | 16.9 | 1961.0 | 1913.0 | 828.0 | 881.0 |
| FAA Hybrid III | 07324-23 | 4 | 21 | 16.0 | 1988.0 | 1900.0 | 840.0 | 885.0 |

| ATD | Belt Type | Goal G | Head Excursion (in) | Left Lap Belt Force | Right Lap Belt Force | | Shoulder Belt Force - Right |
|----------------|--------------|-----------|---------------------------|------------------------|-------------------------|------|-----------------------------------|
| FAA Hybrid III | 2 | 16 | 0.5 | 1.8% | 1.2% | N/A | N/A |
| | | | | | | | |
| FAA Hybrid III | 3 | 21 | 1.6 | 0.8% | 5.9% | 7.4% | N/A |
| | | | | | | | |
| FAA Hybrid III | 4 | 21 | 0.9 | 6.2% | 2.2% | 3.0% | 10.7% |

Comment: Although the sled peak acceleration varied somewhat between the tests, the data was not normalized.

Test Repeatability = $\frac{\text{Maximum Load} - \text{Minimum Load}}{\text{Maximum Load}} * 100\%$

Reference: [4]

Test Repeatability:

Head path: 0.5 - 1.6 inch

Lap belt: 0.8% – 6.2%

Torso strap load: 3.0% – 10.7%



Head Path and Belt Load Comparison - Rigid Seat, Thin Cushion

Hybrid II vs FAA Hybrid III comparison: Head path/ Belt load

| Test Config. | ATD | Test Number | Goal G | Peak G | Head Excursion (in) | Lap Belt Load (lb) | Torso Strap Load (lb) |
|-----------------------|------------|----------------|-----------|-----------|---------------------------|-----------------------|--------------------------|
| | Hybrid II | A99001 | 18 | 18.1 | 39.9 | 2100 | |
| Lap Belt | Tiyblid li | A99002 | 18 | 17.9 | 40.3 | 2164 | NA |
| Horizontal | FAA Hybrid | A98045 | 18 | 18.1 | 40.6 | 2174 | |
| | III | A98046 | 18 | 17.9 | 40.6 | 2208 | |
| | | | | | | | |
| 4 Daint | Hybrid II | A99003 | 16 | 16.3 | 19.5 | 1839 | 1050 |
| 4- Point Restraint | Hybrid II | A99004 | 16 | 16.1 | 19.5 | 1830 | 1000 |
| Horizontal | FAA Hybrid | A99005 | 16 | 16.3 | 19.2 | 1984 | 900 |
| nonzoniai | III | A99006 | 16 | 15.9 | 19.5 | 2068 | 950 |

| Belt Type | Goal G | Head Excursion (in) | Lap Belt Load | Torso Strap Load |
|--------------|-----------|---------------------------|------------------|---------------------|
| 2 | 18 | 0.5 | 2.8% | N/A |
| | | | | |
| 4 | 16 | -0.2 | 10.4% | -9.8% |

Comment: Although the sled peak acceleration varied somewhat between the tests, the data was not normalized.

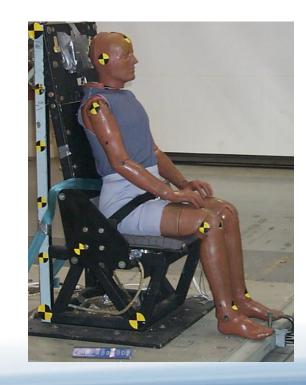
 $Relative \ Error = \frac{Avg(FAA \ Hybrid \ III \ Load) - Avg(Hybrid \ II \ Load)}{Avg \ (Hybrid \ II \ Load)}$

Relative Error:

Head path: -0.2 - 0.5 inch

Lap belt: 2.8% – 10.4%

Torso strap load: -9.8%

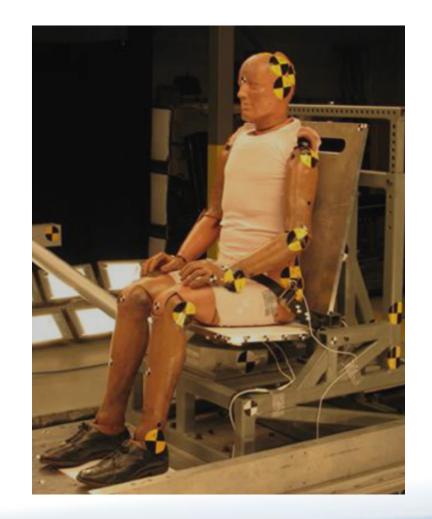


Head Path and Belt Load Comparison - Rigid Seat, No Cushion

Hybrid II Vs FAA Hybrid III Test comparison: Head path/ Belt load

Raw Data

| ATD | Test Number | Belt Type | Goal G | Head Excursio n (in) | Left Lap Belt Force (lb) | Right Lap Belt Force (lb) | Shoulder Belt Force - Left (lb) | Shoulder Belt Force - Right (lb) |
|----------------|----------------|--------------|-----------|----------------------------|--------------------------------|---------------------------------|---------------------------------------|--|
| Hybrid II | 07324-4 | 2 | 16 | 35.2 | 1887.0 | 1870.0 | | |
| Hybrid II | 07324-7 | 2 | 16 | 35.7 | 2002.0 | 1952.0 | | |
| Hybrid II | 07324-28 | 2 | 16 | 36.2 | 1989.0 | 1976.0 | | |
| FAA Hybrid III | 07324-16 | 2 | 16 | 36.9 | 1851.0 | 1776.0 | | |
| FAA Hybrid III | 07324-17 | 2 | 16 | 37.1 | 1837.0 | 1812.0 | | |
| FAA Hybrid III | 07324-18 | 2 | 16 | 37.4 | 1817.0 | 1783.0 | | |
| | | | | | | | | |
| Hybrid II | 07324-8 | 3 | 21 | 18.9 | 1724.0 | 2475.0 | 1606.0 | |
| Hybrid II | 07324-9 | 3 | 21 | 18.9 | 1720.0 | 2509.0 | 1641.0 | |
| Hybrid II | 07324-29 | 3 | 21 | 19.1 | 1599.0 | 2474.0 | 1574.0 | |
| FAA Hybrid III | 07324-20 | 3 | 21 | 19.5 | 1718.0 | 2324.0 | 1712.0 | |
| FAA Hybrid III | 07324-24 | 3 | 21 | 21.1 | 1713.0 | 2420.0 | 1586.0 | |
| FAA Hybrid III | 07324-25 | 3 | 21 | 20.5 | 1727.0 | 2471.0 | 1637.0 | |
| | | | | | | | | |
| Hybrid II | 07324-5 | 4 | 21 | 14.7 | 1904.0 | 1858.0 | 990.0 | 935.0 |
| Hybrid II | 07324-26 | 4 | 21 | 13.4 | 2024.0 | 1958.0 | 920.0 | 995.0 |
| Hybrid II | 07324-27 | 4 | 21 | 13.5 | 2042.0 | 1941.0 | 874.0 | 973.0 |
| FAA Hybrid III | 07324-21 | 4 | 21 | 16.0 | 1865.0 | 1870.0 | 854.0 | 790.0 |
| FAA Hybrid III | 07324-22 | 4 | 21 | 16.9 | 1961.0 | 1913.0 | 828.0 | 881.0 |
| FAA Hybrid III | 07324-23 | 4 | 21 | 16.0 | 1988.0 | 1900.0 | 840.0 | 885.0 |



Reference: [4]

Head Path and Belt Load Comparison - Rigid Seat, No Cushion

Hybrid II Vs FAA Hybrid III Test comparison: Head path/ Belt load

Relative Error

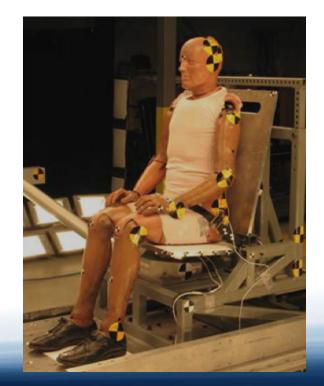
| Belt Type | Goal G | Head Excursion (in) | Left Lap Belt Force | Right Lap Belt Force | Shoulder Belt Force – Left | Shoulder Belt Force - Right |
|-----------|--------|---------------------------|------------------------|-------------------------|----------------------------------|--------------------------------|
| 2 | 16 | 1.4 | -6.3% | -7.3% | N/A | N/A |
| | | | | | | |
| 3 | 21 | 1.4 | 2.3% | -3.2% | 2.4% | N/A |
| | | | | | | |
| 4 | 21 | 2.4 | 2.7% | 1.3% | -9.4% | -12.0% |

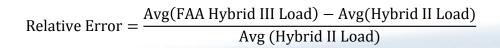
Relative Error:

Head path: 1.4 - 2.4 inch

Lap belt: -7.3% – 2.7%

Torso strap load: -12.0 - 2.4%





Horizontal Test – Real Seat – Hybrid II

Hybrid II repeatability: Head path/ Belt load

Y Belt and No Wall Configuration (Real Seat)

| ATD | Test Number | Belt and Wall Config | | Peak G | dV (ft/s) | X-Path peak (in) | X-Path difference (in) |
|-----------|----------------|-------------------------|----|-----------|--------------|---------------------|------------------------------|
| Hybrid II | A08008 | Y Belt | No | 17.1 | 46.9 | 37.4 | 0.3 |
| Hybrid II | A08009 | Y Belt | No | 17.7 | 47.0 | 37.7 | 0.3 |

Comment: Although the sled peak acceleration varied somewhat between the tests, the total sled velocity change was essentially the same. Therefore the excursion data was not normalized.

Reference: [6,7]

Test Repeatability: 0.3 inch



Head Path Comparison - Real Seat

Hybrid II Vs FAA Hybrid III Test comparison: Head path/ Belt load

Y Belt and No Wall/ near wall Configuration (Real Seat)

Reference: [6,7]

| ATD | Test Number | Belt and Wall Config | | Peak G | dV (ft/s) | X-Path peak (in) | Relative Error (in) |
|----------------|----------------|-------------------------|--------------|--------|--------------|---------------------|---------------------|
| Hybrid II | A09005 | Nylon, Lap Belt | No | 16.5 | 46.6 | 40.7 | 0.6 |
| FAA Hybrid III | A09007 | Nylon, Lap Belt | No | 16.0 | 46.5 | 41.3 | |
| | | | | | | | |
| Hybrid II | A09006 | Poly, Lap Belt | No | 15.8 | 46.3 | 40 | 0.4 |
| FAA Hybrid III | A09008 | Poly, Lap Belt | No | 17.1 | 46.9 | 40.4 | |
| | | | | | | | |
| Hybrid II | A09008 | Y Belt | No | 17.1 | 46.9 | 37.4 | 0.45 (avg) |
| Hybrid II | A09009 | Y Belt | No | 17.7 | 47.0 | 37.7 | |
| FAA Hybrid III | A09009 | Y Belt | No | 17.7 | 47.0 | 38 | |
| | | | | | | | |
| Hybrid II | A09011 | Y Belt | Near Wall | 16.2 | 44.5 | 35.9 | 0.2 |
| FAA Hybrid III | A09011 | Y Belt | Near Wall | 16.2 | 44.5 | 35.7 | |

 $\mbox{Relative Error} = \frac{\mbox{FAA Hybrid III Load} - \mbox{Hybrid II Load}}{\mbox{Hybrid II Load}}$

Relative Error: 0.2 – 0.6 inch



Findings: Head Path During Horizontal Tests - Rigid Seat

Test Repeatability:

Comparison Relative Error:

Lap Belt: Hybrid II = 1 inch FAA Hybrid III = 0.5 inch Lap Belt: 1999 Series: +0.5 inch 2007 Series: +1.4 inch

3-Point Belt: Hybrid II = 0.2 inch FAA Hybrid III = 1.6 inch 3-Point Belt:

2007 Series: +1.4 inch

4-Point Belt: Hybrid II = 1.3 inch FAA Hybrid III = 0.9 inch 4-Point Belt:

1999 Series: -0.2 inch

2007 Series: +2.4 inch

Findings: Head Path During Horizontal Tests – Real Seat

Test Repeatability:

Lap Belt.

Hybrid II = 0.3 inch

Hybrid III = no data available

Comparison Relative Error:

Lap Belt.

Nylon Belt: +0.6 inch

Poly Belt: +0.4 inch

Y-belt: +0.5 inch

Y-belt near wall: -0.2 inch

Findings: Lap Belt Tension – Rigid Seat

Test Repeatability:

Lap Belt.

Hybrid II = 3.0 to 5.7%

FAA Hybrid III = 1.2 to 1.8%

3-Point Belt.

Hybrid II = 1.4 to 7.3%

FAA Hybrid III = 0.8 to 5.9%

4-Point Belt.

Hybrid II = 5.1 to 6.7%

FAA Hybrid III = 2.2 to 6.2%

Comparison Relative Error:

Lap Belt.

1999 Series: +2.8%

2007 Series: -6.3 to -7.3%

3-Point Belt.

2007 Series: -3.2 to +2.3%

4-Point Belt.

1999 Series: +10.4%

2007 Series: +1.3 to +2.7%

Findings: Shoulder Belt Tension – Rigid Seat

Test Repeatability:

3-Point Belt: Hybrid II = 4.1% FAA Hybrid III = 7.4%

4-Point Belt: Hybrid II = 4.8 to 11.7% FAA Hybrid III = 3.0 to 10.7%

Comparison Relative Error:

3-Point Belt.

2007 Series: +2.4%

4-Point Belt.

1999 Series: -9.8%

2007 Series: -9.4 to -12.0%

Conclusions

Forward-facing seats:

Rigid seats:

- Overall, the FAA Hybrid III ATD showed better repeatability than Hybrid II for lumbar load, head path, and belt tension.
- The relative error between the FAA Hybrid III and Hybrid II was generally within the repeatability range of the ATDs.

Real seats:

 Real seat vertical test results showed better repeatability for both ATD's and better relative error than the rigid seat tests. This observation, however, is based on a very small number of real seat tests.

Obliquely-facing seats:

- Large lumbar load differences between the two ATDs were observed.
- Further investigation is needed to determine the source of this difference.

References

- 1). Van Gowdy, Richard DeWeese, Micheal Beebe, Barry Wade, John Duncan, Randy Kelly, James Blaker, "A Lumbar Spine Modification to the Hybrid III ATD for Aircraft Seat Tests", SAE Technical Paper 1999-01-1609, 1999.
- 2). Amanda M Taylor, David M Moorcroft, Richard L Deweese, "Comparison of the Hybrid II, FAA Hybrid III, and THOR-NT in Vertical Impacts", AHS 73rd Annual Forum & Technology Display, Fort Worth, Texas, May 9-17, 2017.
- 3). Joseph Pellettiere, David Moorcroft, Gerardo Olivares, "Anthropomorphic Test Dummy Lumbar Load Variation", The 22nd International Technical Conference on the Enhanced safety of Vehicles, Washington DC, June 13-16, 2011.
- 4). Gerardo Olivares, "Hybrid II and Federal Aviation Administration Hybrid III Anthropomorphic Test Dummy Dynamic Evaluation Test Series". DOT/FAA/AR-11/24, 2011.
- 5). SAE ARP 5765 Rev A, "Analytical Methods for Aircraft Seat Design and Evaluation", SAE International, 2015.
- 6). Richard DeWeese, "Kinematics of Lap Belt Restrained Occupants", Aerospace Structural Impact Dynamics International Conference, Wichita KS, Nov 2012.
- 7). Previously unpublished data from FAA Civil Aerospace Medical Institute test series described in Ref 6.
- 8). H2 and H3 FAA lumbar load response in 14G test, Safron Seats France, Unpublished Case Study, 2019.
- 9). Previously unpublished data from FAA Civil Aerospace Medical Institute follow-on tests related to project described in Ref 1.