# echnic も note technical ite

# Thermal Dynamics of Bomb Calorimeters

Richard E. Lyon

April 2016

DOT/FAA/TC-TN16/16

This document is available to the U.S. public through the National Technical Information Services (NTIS), Springfield, Virginia 22161.

This document is also available from the Federal Aviation Administration William J. Hughes Technical Center at actlibrary.tc.faa.gov.

U.S. Department of Transportation Federal Aviation Administration

#### NOTICE

This document is disseminated under the sponsorship of the U.S. Department of Transportation in the interest of information exchange. The U.S. Government assumes no liability for the contents or use thereof. The U.S. Government does not endorse products or manufacturers. Trade or manufacturers' names appear herein solely because they are considered essential to the objective of this report. The findings and conclusions in this report are those of the author(s) and do not necessarily represent the views of the funding agency. This document does not constitute FAA policy. Consult the FAA sponsoring organization listed on the Technical Documentation page as to its use.

This report is available at the Federal Aviation Administration William J. Hughes Technical Center's Full-Text Technical Reports page: actlibrary.tc.faa.gov in Adobe Acrobat portable document format (PDF).

	1		Technical Report I	Documentation Page	
1. Report No.	2. Government Accession No	).	<ol><li>Recipient's Catalog No.</li></ol>		
DOT/FAA/TC-TN16/16			5 Depart Data		
4. The and Subline			5. Report Date		
THERMAL DYNAMICS OF BOMB CA	LORIMETERS		April 2016	`odo	
			6. Penorming Organization C	Jode	
7. Author(s)			8. Performing Organization R	Report No.	
Richard E. Lyon					
Fire Safety Branch					
William J. Hughes Technical Center					
Atlantic City International Airport, NJ 08	405				
9. Performing Organization Name and Address			10. Work Unit No. (TRAIS)		
Federal Aviation Administration					
Fife Safety Branch ANG-E21 William I. Hughes Technical Center			11. Contract or Grant No.		
Atlantic City International Airport, NJ 08	405				
12. Sponsoring Agency Name and Address			13. Type of Report and Peric	d Covered	
U.S. Department of Transportation/Federa	al Aviation Administra	tion			
Northwest Mountain Regional Office					
1601 Lind Avenue, SW					
Renton, WA 98057					
			14. Sponsoring Agency Code	9	
			ANM-115		
15. Supplementary Notes			·		
16. Abstract					
The validated solution to a two-ten	rm heat transfer mo	odel of a bomb cale	orimeter allows dir	rect calculation of	
the heat released in an arbitrary pa	rocess from the red	corded temperature	history without the	ne need to correct	
for non-adiabatic behavior. The h	eat transfer coeffic	cients and thermal	capacities of the	bomb calorimeter	
used in the heat calculation are de	etermined parametr	ically from the ter	nperature response	to a known heat	
impulse (i.e., benzoic acid combu	stion). This metho	dology allows acc	urate measuremen	t of heat released	
intermittently or during an exten	ded period of tim	e in a bomb calor	rimeter, as occurs	during electrical	
resistance heating and subsequent thermal runaway of lithium ion batteries.					
17 Kou Mordo		10 Distribution Statement			
17. Key words					
Lithium ion batteries, Thermal runaway, Internal heating, Thermal energy Mass loss Computing Bomb calorimeter		This document is available to the U.S. public through the			
		Virginia 22161. This document is also available from the			
		Federal Aviation Ad	lministration William	J. Hughes Technical	
40. Consulty Observity (at this second)		Center at actlibrary.to	c.faa.gov.		
	20. Security Classif. (of this p	age)	21. NO. OF Pages	ZZ. Price	
Unclassified	Unclassified		20		

Reproduction of completed page authorized

## ACKNOWLEDGEMENTS

The author is grateful to Dr. Jay Albert of Parr Instruments for providing data from the isoperibol calorimeter, to Dr. Richard N. Walters of the FAA for conducting the combustion experiments in the static jacket calorimeter, and to Professor James G. Quintiere of the University of Maryland for engaging in helpful discussions. Certain commercial equipment, instruments, materials, and companies are identified in this paper to adequately specify the experimental procedure. This in no way implies endorsement or recommendation by the Federal Aviation Administration.

# TABLE OF CONTENTS

	Page
EXECUTIVE SUMMARY	vii
INTRODUCTION	1
CALORIMETER MODEL	1
Heat Released in an Arbitrary Process Temperature History for a Heat Pulse (Combustion)	2 4
MODEL VALIDATION	5
Adiabatic Method ( $K_2 = 0$ ) Isoperibol Method ( $\theta_J = \text{Constant} > 0$ ) Static Jacket Calorimeter ( $\theta_J \approx 0$ )	5 5 7
CONCLUSIONS	11
REFERENCES	11

# LIST OF FIGURES

Figure		Page
1	The bomb calorimeter system of lumped heat transfer analysis	2
2	Plot of equation 18 used to obtain $K_2$ for isoperibol bomb calorimeter	6
3	Measured and calculated temperature history for $Q_0 = 23,028$ J heat impulse in isoperibol calorimeter	6
4	Measured temperature rise and calculated heat for benzoic acid combustion $Q_0 = 23,028$ J in isoperibol bomb calorimeter	7
5	Measured and calculated temperature histories of water bath for $Q$ in table 3	8
6	Temperature histories in the vicinity of $\Delta T_{\text{max}}$ for $Q_0$ in table 3	10
7	Calculated <i>Q</i> using equation 21 and temperature histories of figure 5 compared to nominal values $Q_0 = 14.63$ kJ and $Q_0 = 26.165$ kJ in table 3	11

# LIST OF TABLES

Table		Page
1	Heat of combustion $(Q)$ and thermal constants of isoperibol bomb calorimeter	6
2	Static jacket calorimeter component masses and heat capacities	8
3	Best-fit thermal response parameters for static jacket bomb calorimeter	8

#### EXECUTIVE SUMMARY

An analysis was performed to account for instrument dynamics and heat losses or gains at the boundaries on the heat measured in bomb calorimeters by different standard methods. The objective was to develop a general relationship between the measured temperature history and the heat released in a bomb calorimeter in a prolonged process during which the thermal dynamics of the calorimeter and heat losses to the environment could affect the result. An analytic expression was derived and tested by calculating the heat of combustion from the measured temperature history for the three common methods of bomb calorimetry: adiabatic, isoperibol, and static jacket. In each case, the analytic expression gave accurate heats of combustion by explicitly accounting for the instrument dynamics and heat losses at the boundaries without the need for empirical temperature corrections as in the standard methods. Therefore, the validated analytic result is useful for calculating the heat of an intermittent or prolonged process, such as electrical resistance heating and subsequent thermal runaway of lithium ion batteries, in a bomb calorimeter.

#### INTRODUCTION

The bomb calorimeter [1] is widely used to determine the combustion heat of foods and condensed fuels using standard methods [2–7]. These standard methods use elaborate piecewise corrections to account for non-adiabatic conditions (i.e., heat exchange with the surroundings during a test in which heat is released instantaneously, as an impulse) [8]. If heat is released intermittently or over an extended time period, it is not clear that the mathematics used in the standard methods to compute the heat of an impulsive process are correct. The three principle methods of bomb calorimetry (adiabatic, isoperibol, and static jacket) differ only in their boundary conditions. For each of these methods, the heat of an arbitrary process occurring in the pressure vessel should be calculable from the temperature history of the water bath. The present approach is to use a lumped heat transfer model of a standard bomb calorimeter. Good agreement between the theoretical and measured calorimeter responses to a heat pulse (combustion) would validate the model and provide the heat transfer and thermal capacitance coefficients for the bomb calorimeter.

#### CALORIMETER MODEL

The thermal dynamics of bomb calorimeters are modeled using a lumped heat transfer analysis in which heat is released in a pressure vessel/bomb immersed in a stirred water bath that is surrounded by a static air space bounded by an insulated (static) jacket, a constant/controlled temperature jacket (isoperibol), or a changing temperature (adiabatic) jacket. A schematic diagram of a standard bomb calorimeter system used for combustion measurements [2–4] is shown in figure 1. The steel pressure vessel (bomb) containing the sample at a uniform temperature  $T_b$  is immersed in a stirred water bath at temperature  $T_w$ , where the temperature of the calorimeter system is measured and recorded as a function of time, *t*. The water bath is inserted into an enclosed jacket having temperature  $T_j$  that is either static ( $T_j \approx T_{\infty}$ ), held at a constant temperature ( $T_j \ge T_{\infty} = \text{constant}$ ), or matched to the water bath temperature to prevent heat escape ( $T_j = T_w$ ). These three boundary conditions (methods) are referred to as static jacket, isoperibol, and adiabatic bomb calorimetry, respectively. When heat *Q* is generated inside the pressure vessel (bomb) during an experiment at rate dQ/dt = Q', the rate of change of the bomb's internal energy is:

$$C_{1}\frac{dT_{b}}{dt} = Q' - K_{1}(T_{b} - T_{w})$$
(1)

The rate of internal energy change of the water bath is the difference between the heat transferred to the water bath from the bomb and the heat lost from the water bath to the surroundings through the air gap and jacket:

$$C_2 \frac{dT_w}{dt} = K_1 (T_b - T_w) - K_2 (T_w - T_j)$$
(2)

Temperature measurements are made in the water bath during the experiment, and the heat of the process is calculated from the temperature difference,  $\theta_2 = T_w T_i$ . In the bomb calorimeter of

figure 1, depending on the method used, the water bath either: 1) retains all the heat of the process (adiabatic), 2) absorbs heat from the combustion process and jacket (isoperibol), or 3) absorbs heat from the combustion process and loses heat to the environment (static) jacket. The jacket temperature,  $T_j$ , which is a boundary condition, defines these three methods of bomb calorimetry. They are:  $T_w < T_j$  (isoperibol);  $T_w = T_j$  (adiabatic),  $T_w > T_j$  (static jacket).



Figure 1. The bomb calorimeter system of lumped heat transfer analysis

#### HEAT RELEASED IN AN ARBITRARY PROCESS

To begin, define a temperature difference between the pressure vessel/bomb and the water bath,  $\theta_1 = T_b - T_w$  and a characteristic time for the bomb,  $\tau_1 = C_1/K_1$ , such that equation 1 becomes:

$$\frac{d\theta_1}{dt} + \frac{\theta_1}{\tau_1} = \frac{Q'}{C_1} \tag{3}$$

The temperature difference between the water bath and the jacket is  $\theta_2 = (T_w - T_j)$  and the characteristic time of the calorimeter system (bomb + water bath) is  $\tau_2 = C_2/K_2$ . With these definitions, equation 2 becomes:

$$\frac{d\theta_2}{dt} + \frac{\theta_2}{\tau_2} = \frac{K_1}{C_2}\theta_1 \tag{4}$$

Equations 3 and 4 are coupled differential equations that must be solved for the change in the measured bath temperature history,  $T_w(t)$ . The strategy is to solve equations 3 and 4 for the heat of an arbitrary process in terms of  $\theta_2(t)$ . Taking the Laplace transforms of equation 3, with superscripted bars indicating the transformed function and *s* indicating the transform variable:

$$s\overline{\theta}_1 - \theta_1(0) + \frac{\overline{\theta}_1}{\tau_1} = \frac{s\overline{Q}}{C_1} - Q(0)$$
(5)

Because  $\theta_1(0) = T_b - T_w = Q(0) = 0$  at t = 0:

$$\overline{\theta}_{1} = \frac{s\overline{Q}}{C_{1}} \left( \frac{\tau_{1}}{1 + s\tau_{1}} \right)$$
(6)

The Laplace transform of equation 4 is:

$$s\overline{\theta}_2 - \theta_2(0) + \frac{\overline{\theta}_2}{\tau_2} = \frac{K_1}{C_2}\overline{\theta}_1$$
(7)

Substituting equation 6 for  $\overline{\theta}_1$  into equation 7 and rearranging terms, the transformed equation for the heat of a process is:

$$\overline{Q} = K_2 \frac{\overline{\theta}_2}{s} + C_2 (1 + \tau_1 / \tau_2) \overline{\theta}_2 + C_2 \tau_1 s \overline{\theta}_2 - C_2 \theta_2 (0) \left(\frac{1}{s} + \tau_1\right)$$
(8)

The inverse Laplace transform of equation 8, with x as the dummy time variable of integration, is:

$$Q(t) = K_2 \int_0^t \theta_2(x) dx + C_2 \left(1 + \frac{\tau_1}{\tau_2}\right) \theta_2(t) + C_2 \tau_1 \frac{d\theta_2(t)}{dt} - C_2 \theta_2(0) + C_2 \tau_1 \delta(0)$$
(9)

The last term in equation 9 contains the unit impulse,  $\delta(t)$ . For a heat pulse at t = 0:

$$\delta(0) = \begin{cases} 0 & t \neq 0 \\ \infty & t = 0 \end{cases}$$

If the initial condition is  $T_b = T_w = T_0$  when the bomb is fired at t = 0, then,  $\theta_2(t) = (T_w - T_j) = (T_w - T_0) - (T_j - T_0) = \Delta T - \theta_j$  and  $\theta_2(0) = -\theta_j$ . With these substitutions and with  $\delta(0) = 0$ , equation 9 becomes:

$$Q(t) = K_2 \int_0^t \Delta T(x) dx + C_2 \left(1 + \frac{\tau_1}{\tau_2}\right) \Delta T(t) + C_2 \tau_1 \frac{d\Delta T(t)}{dt}$$

$$- K_2 \int_0^t \theta_j(x) dx - C_2 \tau_1 \frac{d\theta_j}{dt} - C_2 \left(1 + \frac{\tau_1}{\tau_2}\right) \theta_j + C_2 \theta_j$$
(10)

For constant  $\theta_i$  and for  $\tau_2 \gg \tau_1$  the last three terms vanish:

$$Q(t) = K_2 \int_{0}^{t} \Delta T(x) dx + C_2 \left( 1 + \frac{\tau_1}{\tau_2} \right) \Delta T(t) + C_2 \tau_1 \frac{d\Delta T(t)}{dt} - K_2 \int_{0}^{t} \theta_j(x) dx$$
$$= C_2 \left\{ \frac{1}{\tau_2} \int_{0}^{t} \Delta T(x) dx + \left( 1 + \frac{\tau_1}{\tau_2} \right) \Delta T(t) + \tau_1 \frac{d\Delta T(t)}{dt} - \frac{1}{\tau_2} \int_{0}^{t} \theta_j(x) dx \right\}$$
(11)

 $=C_2f(t)$ 

Equation 11 shows that the heat of any process in a bomb calorimeter is proportional to the thermal capacity of the calorimeter system,  $C_2$ , and a function of time, f(t). Equation 11 is useful for direct calculation of the heat of process (including combustion) in an adiabatic, isoperibol, or static jacket bomb calorimeter without resorting to complex temperature corrections [8–10] for non-adiabatic boundary conditions.

#### TEMPERATURE HISTORY FOR A HEAT PULSE (COMBUSTION)

When heat is generated rapidly in the calorimeter at time t = 0 (e.g., by combustion) such that the duration of heat generation is much shorter than  $\tau_1$  or  $\tau_2$ , the heat release rate Q' can be written in terms of the heat pulse of magnitude Q released at t = 0 using the unit impulse  $\delta(t)$ . Substituting  $Q'(t) = Q\delta(0)$  into equation 3, taking Laplace transforms, and rearranging:

$$\overline{\theta}_1 = \frac{Q}{C_1} \left( \frac{\tau_1}{1 + s \tau_1} \right) + \theta_1(0) = \frac{Q}{C_1} \left( \frac{\tau_1}{1 + s \tau_1} \right)$$
(12)

Substituting equation 11 for  $\overline{\theta}_1$  into equation 7, solving for  $\overline{\theta}_2$  and taking the inverse Laplace transform of the result:

$$\theta_2(t) = \theta_2(0)e^{-t/\tau_2} + \frac{Q}{C_2(1-\tau_1/\tau_2)} \left( e^{-t/\tau_2} - e^{-t/\tau_1} \right)$$
(13)

Substituting  $\theta_2(0) = -\theta_j$  into equation 12 gives the temperature history of a bomb calorimeter for a heat pulse of magnitude *Q* released at *t* = 0:

$$\Delta T = \theta_j \left( 1 - e^{-t/\tau_2} \right) + \frac{Q}{C_2 (1 - \tau_1 / \tau_2)} \left( e^{-t/\tau_2} - e^{-t/\tau_1} \right)$$
(14)

A fit of equation 14 to the measured temperature history  $\Delta T(t)$  for a heat pulse of known magnitude *Q* allows the calorimeter constants  $\tau_1$ ,  $\tau_2$ , and  $C_2$  to be determined.

#### MODEL VALIDATION

#### <u>ADIABATIC METHOD ( $K_2 = 0$ )</u>

For the adiabatic calorimeter, no heat is lost to the surroundings. This is equivalent to setting  $K_2 = 0$  and  $\tau_2 = \infty$  in equations 11 and 14. Making these substitutions in equation 14 gives the temperature history for an adiabatic calorimeter:

$$\Delta T(t) = \frac{Q}{C_2} \left( 1 - e^{-t/\tau_1} \right)$$
(15)

Equation 11 gives the heat of a process in an adiabatic bomb calorimeter in terms of the measured temperature rise of the bath when  $K_2 = 0$  and  $\tau_2 = \infty$ :

$$Q(t) = C_2 \left( \Delta T + \tau_1 \frac{d\Delta T}{dt} \right)$$
(16)

When the bomb and water bath reach equilibrium,  $d\Delta T/dt = 0$ , and the total heat of the process is proportional to the steady-state (maximum) temperature rise:

$$Q = Q(\infty) = C_2 \Delta T(\infty) = C_2 \Delta T_{\text{max}}$$
(17)

## **ISOPERIBOL METHOD** ( $\theta_{I} = \text{CONSTANT} > 0$ )

In isoperibol bomb calorimetry, the jacket is maintained at a constant temperature that is above the bath temperature, so heat is added to the calorimeter before (t < 0), during (t = 0), and after (t > 0) firing. If the temperature rise of the water bath prior to firing (also called the initial period [8]) is measured, the rate of heat transfer from the jacket to the calorimeter is calculable from equation 14 with Q = 0. The result is:

$$\frac{T_j - T_0}{T_j - T_w} = e^{t/\tau_2}$$
(18)

Consequently, a plot of  $\ln[(T_j - T_0)/(T_j - T_w)] = -\theta_j/\theta_2(t)$  versus 1/t has slope  $1/\tau_2 = K_2/C_2$  from which  $K_2$  can be determined if  $C_2$  is known. Figure 2 shows such a plot for data obtained from the literature [11] and private communications [12] for an isoperibol bomb calorimeter. For this experiment, the reported [12] experimental parameters are  $T_j = 25.58^{\circ}$ C (298.58K) and  $Q_0 = 23,028$  J by benzoic acid combustion, and the calorimeter constants are:  $C_2(1 + \tau_1/\tau_2) = 9965$  J/K. From this information and the slope of the plot in figure 2,  $K_2 = (4.03 \times 10^{-5} \text{ s}^{-1})(9946 \text{ J/K}) = 0.40 \text{ W/K}$ , which is the reported value for the calorimeter [12].



Figure 2. Plot of equation 18 used to obtain  $K_2$  for isoperibol bomb calorimeter

Figure 3 is a plot of the temperature history of an isoperibol bomb calorimeter having  $T_j = 25.58$ °C (298.6K) [11] and the fit of these data to equation 14 for  $Q_0 = 23,028$  J using  $\tau_1 = 0.8$  minutes and  $\tau_2 = 415$  minutes as fitting parameters from which  $C_2(1 + \tau_1/\tau_2) = 9965$  J/K and  $C_2 = (9965 \text{ J/K})/(1 + 0.8/415) = 9946$  J/K. These parameters are summarized in table 1.

Table 1. Heat of combustion (Q) and thermal constants of isoperibol bomb calorimeter



Figure 3. Measured (points) and calculated (line) temperature history for  $Q_0 = 23,028$  J heat impulse in isoperibol calorimeter

In isoperibol calorimetry, the jacket temperature is held constant so  $\theta_j = T_j - T_0$  is also a constant, and equation 11 becomes:

$$Q(t) = K_2 \int_0^t \Delta T(x) dx + C_2 \left(1 + \frac{\tau_1}{\tau_2}\right) \Delta T(t) + C_2 \tau_1 \frac{d\Delta T(t)}{dt} - K_2 \theta_j t$$
(19)

Figure 4 is a locally weighted least-squares fit of equation 19, evaluated for the temperature history in figure 3 and a benzoic acid heat of combustion  $Q_0 = 23,028$  J. The steady-state value at t > 7 minutes is  $\langle Q \rangle = 23,002 \pm 3$ J, which is within 0.1% of the reported  $Q_0$  in table 1.



Figure 4. Measured temperature rise (points) and calculated heat (line) for benzoic acid combustion  $Q_0 = 23,028$  J in isoperibol bomb calorimeter

#### <u>STATIC JACKET CALORIMETER ( $\theta_{J} \approx 0$ )</u>

A static jacket bomb calorimeter (Model 1341, Plain Jacket Oxygen Bomb Calorimeter, Parr Instrument Company, Moline, IL) was used in our laboratory for these experiments according to standard procedures [2-4]. The masses and heat capacities of the calorimeter system components are given in table 2. Benzoic acid ( $Q_0 = 26.453 \text{ kJ/g}$ ) from a commercial source was used for calibration and stored in a dessicator until use. Ambient ( $T_{\infty}$ ) and water bath temperatures ( $T_w$ ) were logged continuously at 1 Hz for the duration of the experiment using a commercial data acquisition module and software (National Instruments, 11500 Mopac Expressway, Austin, TX 78759).

Calorimator Component	Mass	Specific	Heat
Calofiniteter Component	(g)	(J/g-K)	(J/K)
Combustion Bomb	2973	0.46	1368
Calorimeter Vessel	847	0.46	390
Bath Water	2000	4.18	8360
	10,118		

 Table 2. Static jacket calorimeter component masses and heat capacities

Figure 5 is a plot of the measured temperature rise histories of the water bath  $\Delta T(t)$  for benzoic acid combustions (including the fuse corrections),  $Q_0 = 14,640$  J and 26,194 J. The theoretical temperature rise history (equation 14 with  $\theta_j = T_{\infty} - T_0$ ) was fit to the experimental data for each  $Q_0$ , resulting in the solid line in figure 5 and the best-fit parameters in table 3. Note the similarity of the parametrically determined thermal response parameters in tables 1, 2, and 3 for bomb calorimeters from the same manufacturer. In tables 2 and 3,  $C_1 = 1368$  J/K and  $\tau_1 = 60$ s, so the heat transfer coefficient between the pressure vessel and the water bath is  $K_1 = C_1/\tau_1 = 22.8$  W/K = 1.37 kJ/min-K, or approximately 500 W/m<sup>2</sup>-K, based on the surface area of the bomb.

 Table 3. Best-fit thermal response parameters for static jacket bomb calorimeter



Figure 5. Measured and calculated temperature histories of water bath for Q in table 3

The static jacket bomb calorimeter can be classified as a ballistic calorimeter in that the temperature history for a heat pulse follows a ballistic trajectory, having a rising, maximum, and descending period [13]. As with the adiabatic and isoperibol methods of bomb calorimetry, the heat of a generalized process in a static jacket bomb calorimeter is calculable from the experimental data using equation 11, and the temperature history for a combustion experiment is given in equation 14. However, in this case, ,  $T_j \approx T_{\infty}$ , and the boundary condition is quasi-static,  $\theta_j = T_{\infty} - T_0$ .

Figure 6 is an expanded plot of the first several minutes of data in figure 5. The theoretical temperature histories calculated from equation 14 and the thermal response parameters in table 3 are shown as the solid lines through the experimental data points. The temperature in figure 6 reaches a maximum  $\Delta T_{\text{max}}$  in the static jacket bomb calorimeter at approximately 6 to 7 minutes after firing (i.e.,  $t_{\text{max}} \approx 6-7$  minutes). The theoretical value of  $t_{\text{max}}$  is obtained by substituting equation 14 into the integral term of equation 11, with  $\theta_j \approx 0$  for the static jacket, and solving for the upper limit  $t_{\text{max}}$  at  $\Delta T_{\text{max}}$  when  $d\Delta T/dt = 0$ . The result, which is evaluated using  $\tau_1$  and  $\tau_2$  from table 3, is:

$$t_{\max} = \frac{\tau_1 \tau_2}{\tau_2 - \tau_1} \ln \left[ \frac{\tau_2}{\tau_1} \right] = 6.3 \text{ minutes}$$
(20)

This theoretical result, which is in excellent agreement with the data in figure 6, shows that the time to the maximum temperature rise for a combustion experiment in a static jacket bomb calorimeter,  $t_{\text{max}}$ , depends only on the time constants  $\tau_1$  and  $\tau_2$  of the bomb and water bath, respectively. Substituting equation 20 for the upper limit of the integral term in equation 11 when  $d\Delta T/dt = 0$  and  $x = \tau_1/\tau_2$ , shows:

$$Q = C_2 \left\{ 1 + x - \frac{x \ln[x]}{1 - x} \right\} \Delta T_{\max} = C \Delta T_{\max}$$
(21)



Figure 6. Temperature histories in the vicinity of  $\Delta T_{\text{max}}$  for  $Q_0$  in table 3

Equation 21 shows that the energy equivalent of the calorimeter  $C = Q/\Delta T_{\text{max}}$  is a constant that depends only on the thermal response times of the bomb ( $\tau_1$ ) and water bath ( $\tau_2$ ). According to the present analysis, all thermal leakage and response time corrections for a static jacket bomb calorimeter are contained in the empirical proportionality constant *C*, and adjustments to  $\Delta T_{\text{max}}$  are unnecessary and inaccurate. Note that when  $\tau_2 = \infty$  and x = 0, equation 21 is the adiabatic result,  $Q = C_2 \Delta T_{\text{max}}$ .

In static jacket bomb calorimetry, the jacket temperature  $T_j$  approximates the time-average ambient temperature,  $\langle T_{\infty} \rangle$ . In this case,  $\theta_j$  will be a small constant approaching zero,  $\theta_{j,\infty} = T_j - T_0 \approx \langle T_{\infty} \rangle - T_0$ , and equation 11 becomes:

$$Q(t) = K_2 \int_{0}^{t} \Delta T(x) dx + C_2 \left( 1 + \frac{\tau_1}{\tau_2} \right) \Delta T(t) + C_2 \tau_1 \frac{d\Delta T(t)}{dt} - K_2 \theta_{j,\infty} t$$
(22)

Figure 7 is a locally weighted, least-squares fit of equation 22, evaluated for the data of figure 5 with the thermal parameters in table 3. The average plateau values between the vertical dashed lines in figure 7,  $\langle Q \rangle = 14,632$  J and  $\langle Q \rangle = 26,165$  J, are in excellent agreement with the nominal  $Q_0$  in table 3 for the benzoic acid calibration experiments.



Figure 7. Calculated *Q* using equation 21 and temperature histories of figure 5 compared to nominal values  $Q_0 = 14.63$  kJ and  $Q_0 = 26.165$  kJ in table 3

#### CONCLUSIONS

The heat released in an arbitrary process in a bomb calorimeter can be computed from the measured temperature history for three standard methods: adiabatic, isoperibol, and static jacket. For a heat impulse such as combustion, the heat released in adiabatic and static jacket calorimeters is proportional to the maximum temperature according to standard methods. For intermittent or protracted heat release histories, as occurs during electrical resistance heating and subsequent thermal runaway of lithium ion batteries, the heat released is directly calculable from the recorded temperature history using the present methodology without the need for semi-empirical corrections for non-adiabatic behavior.

#### REFERENCES

- 1. Handbook of Thermal Analysis and Calorimetry, Volume 1, Principles and Practice, Combustion Calorimeters, Elsevier M.E. Brown, Ed., Amsterdam, 1998, Chapter 14.10, pp. 657–669.
- 2. "Solid Mineral Fuels: Determination of Gross Calorific Value by the Bomb Calorimetric Method and Calculation of Net Calorific Value," ISO 1928, International Standards Organization, Geneva, Switzerland.
- 3. Solid Recovered Fuels. Determination of Calorific Value, British Standard European Norm, BS EN 15400.
- 4. Solid Biofuels. Determination of Calorific Value, British Standard European Norm, BS EN 14918.
- 5. ASTM D 240, "Test Method for Heat of Combustion of Liquid Hydrocarbon Fuels by Bomb Calorimeter," ASTM International, West Conshohocken, Pennsylvania.

- 6. ASTM D 4809, "Test Method for Heat of Combustion of Liquid Hydrocarbon Fuels by Bomb Calorimeter (Precision Method)," ASTM International, West Conshohocken, Pennsylvania.
- 7. ASTM D 5865, "Standard Test Method for Gross Calorific Value of Coal and Coke," ASTM International, West Conshohocken, Pennsylvania.
- 8. Santos, L., Silva, M., Schroder B., and Gomes, L., "Labtermo: Methodologies for the Calculation of the Corrected Temperature Rise in Isoperibol Calorimeters," *Journal of Thermal Analysis and Calorimetry*, 89, 1,2007, pp 175–180.
- 9. King, A. and Grover, H., "Temperature Correction Methods in Calorimetry," *Journal of Applied Physics*, 12, 7, 1941, pp. 557–569.
- 10. West, E.D. and Churney, K.I., "A Two Body Model for Calorimeters With Constant Temperature Environments," *Journal of Applied Physics*, 39, 9, 1968, pp. 4206–4215.
- 11. "Digital Thermometer 6775/6775A" *Operating Instruction Manual 545M*, Parr Instrument Company, 2013, Moline, Illinois, 61265.
- 12. Private Communication, Dr. Jay Albert, Technical Director, Parr Instrument Company, Moline, Illinois, 61265.
- 13. Miller, D.S. and Payne, P.R., "A Ballistic Bomb Calorimeter," *British Journal of Nutrition*, Vol. 13, 1959, pp. 501–508.