

Engine Nacelle Halon Replacement



Federal Aviation
Administration

Presented to: International Aircraft Systems Fire
Protection Working Group

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Presentation Overview

- **Discuss the “MPSe rev03 to rev04” transition**
 - Overview
 - Status

- **Flow visualization, Small-scale wind tunnel (SSWT) activity**

MPSe Rev 03 → 04, Overview

- **Issues driving the test process revision**
 - Terminate the use of halon 1301 in the process
 - Minimize the impact of agent injection on determining the “equivalent” agent quantity

MPSe Rev 03 → 04, Overview

Terminating Halon 1301 Usage

- **Modify halon benchmark process**
 - Use surrogate to replicate halon 1301 fire suppression behavior
 - Surrogate = HFC-125
 - Revise benchmark process
- **Begin characterizing fire threats**
 - Utilize instrumentation in the test article for characterization
 - Benchmark process will likely be dropped in the future

MPSe Rev 03 → 04, Overview

Minimizing the Effect of Injecting Fire Extinguishant

- **Modify test process**

- Change from an iterative search to a proof-test
 - deliver agent in pre-determined quantity (amount x residence time)
 - predetermined quantity based on accepted bench-scale test
 - based on cup-burner assay
 - agent quantity = (factor-of-safety) x (cup-burner finding)
 - residence time = 0.5 sec
- Requires preliminary testing to produce the agent distribution in the test article for 2 air flow conditions

MPSe Rev 03 → 04, Overview

Minimizing the Effect of Injecting Fire Extinguishant

- **Review agent measurement method**

- Conduct flow observations

- smoke visualization (currently on-going)
 - SSWT & 2 aerodynamic models
 - nacelle fire simulator (NFS)
- gas distribution behavior in the NFS

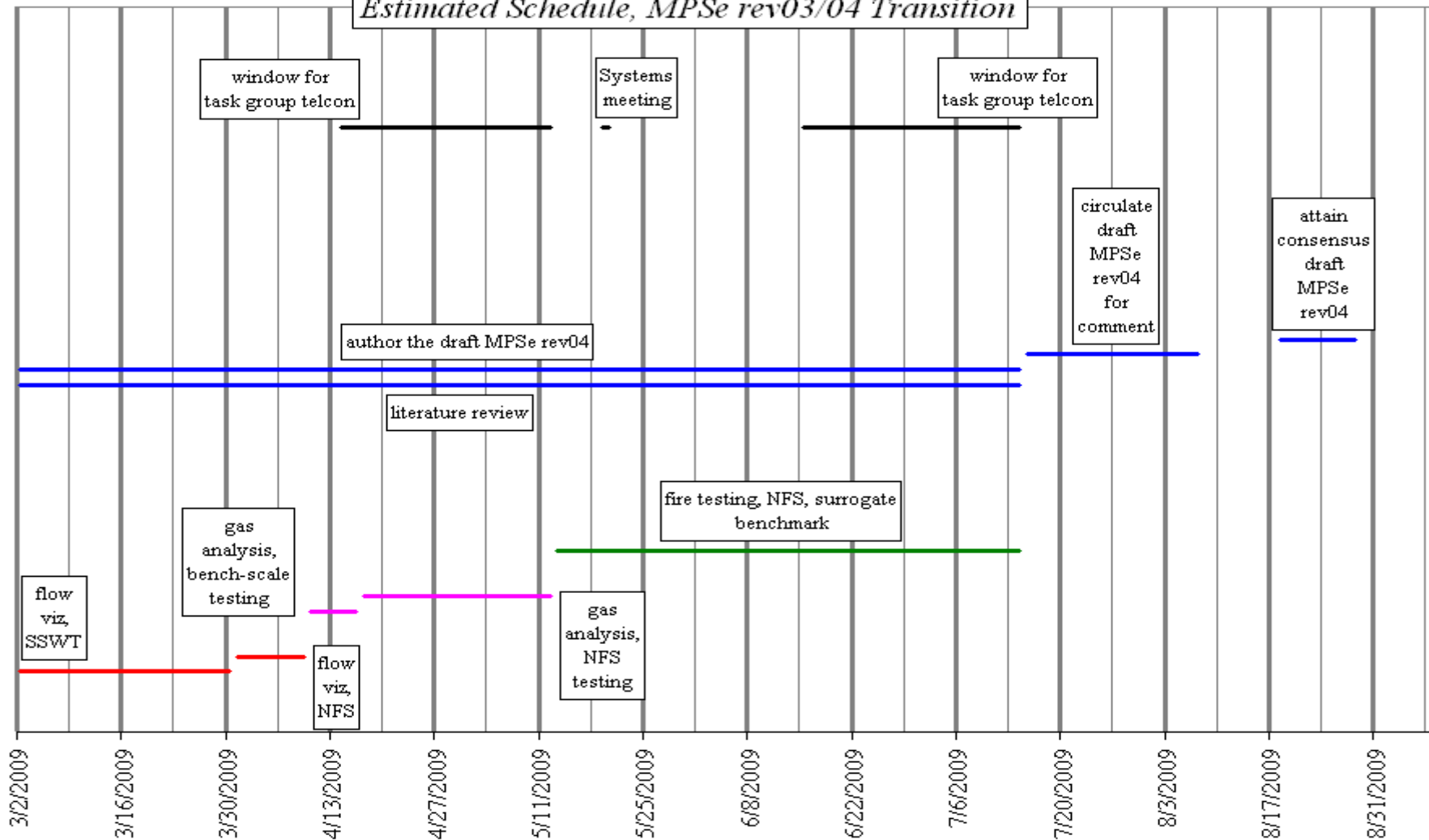
- Revise measurement methods

- as needed...
- based on indications from observations & published literature

MPSe Rev 03 → 04, Status

Original Schedule

Estimated Schedule, MPSe rev03/04 Transition



MPSeRev04-sched.xls, sched LOOSE 2009313 c

MPSe Rev 03 → 04, Status

Current Status

- **Flow observations near-complete**
 - Task group telephone conference call expected shortly
 - discuss flow observations
 - late May/early June
- **Remaining tasks are pending**

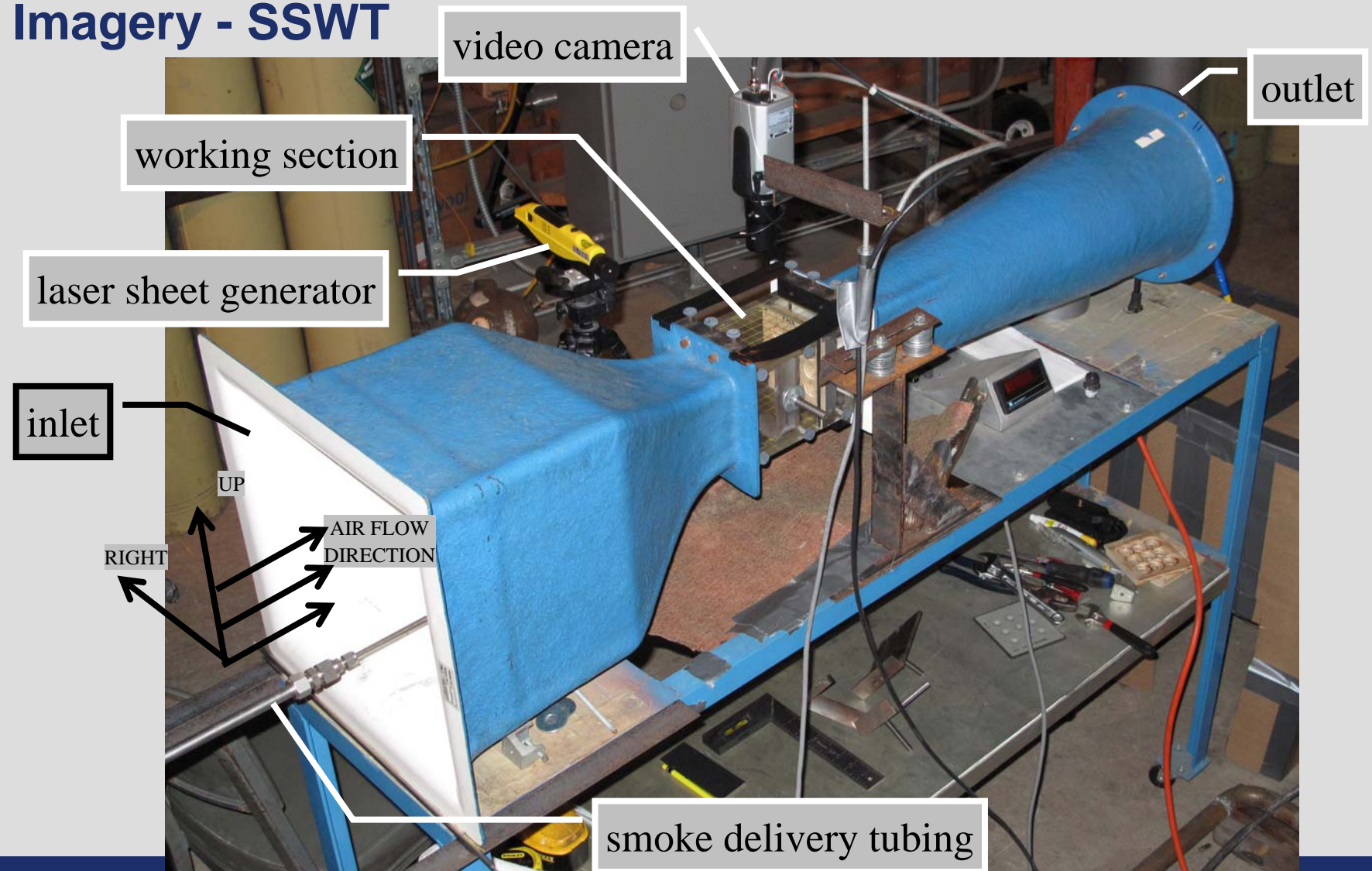
Flow visualization, SSWT Activity

Overview

- **Utilizing SSWT to visualize wake regions**
 - Wake regions are related to flame-holders in the NFS
 - Will use indications for placing gas sample points in the NFS
- **SSWT details**
 - Suction tunnel
 - driven by 0-90 VDC motor
 - speeds up to 50 ft/sec (15.2 m/sec)
 - Working section 4 x 4 x 7.5 inches (102 x 102 x 191 mm)
 - Two aerodynamic models; tube array & “fuel pan”
 - Delivering smoke to visualize flow near models
 - Red laser sheet illuminates horizontal planes

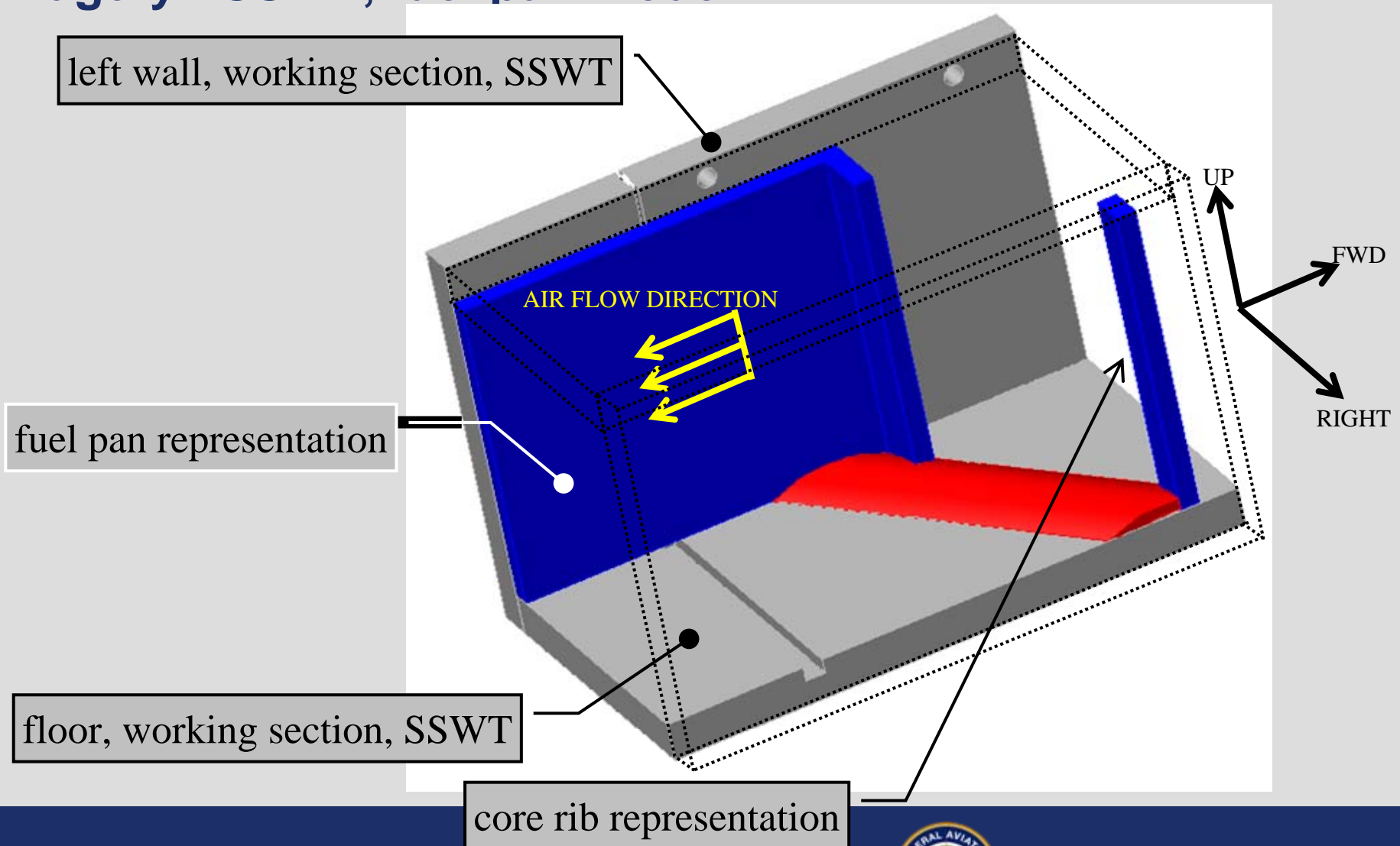
Flow visualization, SSWT Activity

Imagery - SSWT



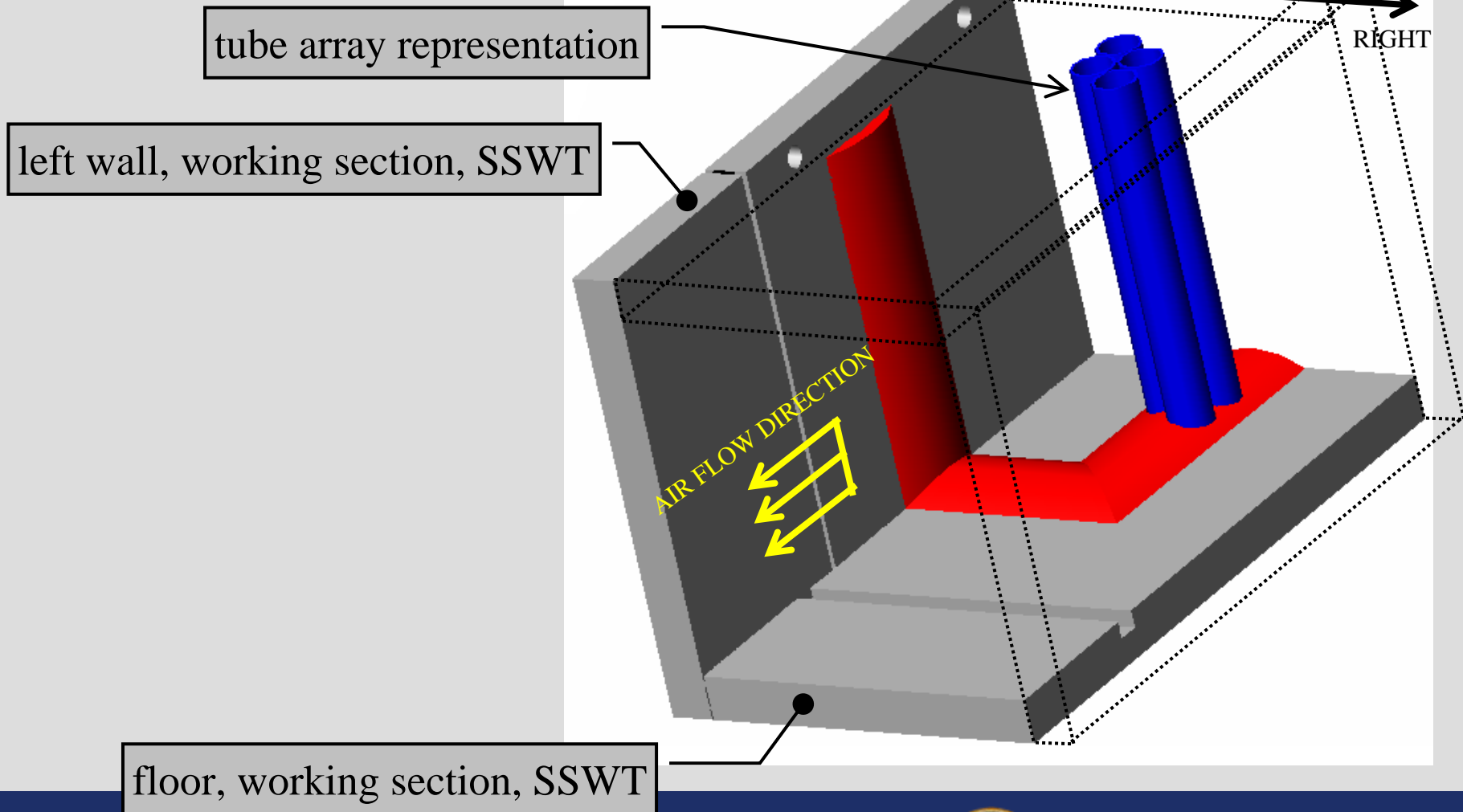
Flow visualization, SSWT Activity

Imagery - SSWT, fuel pan model



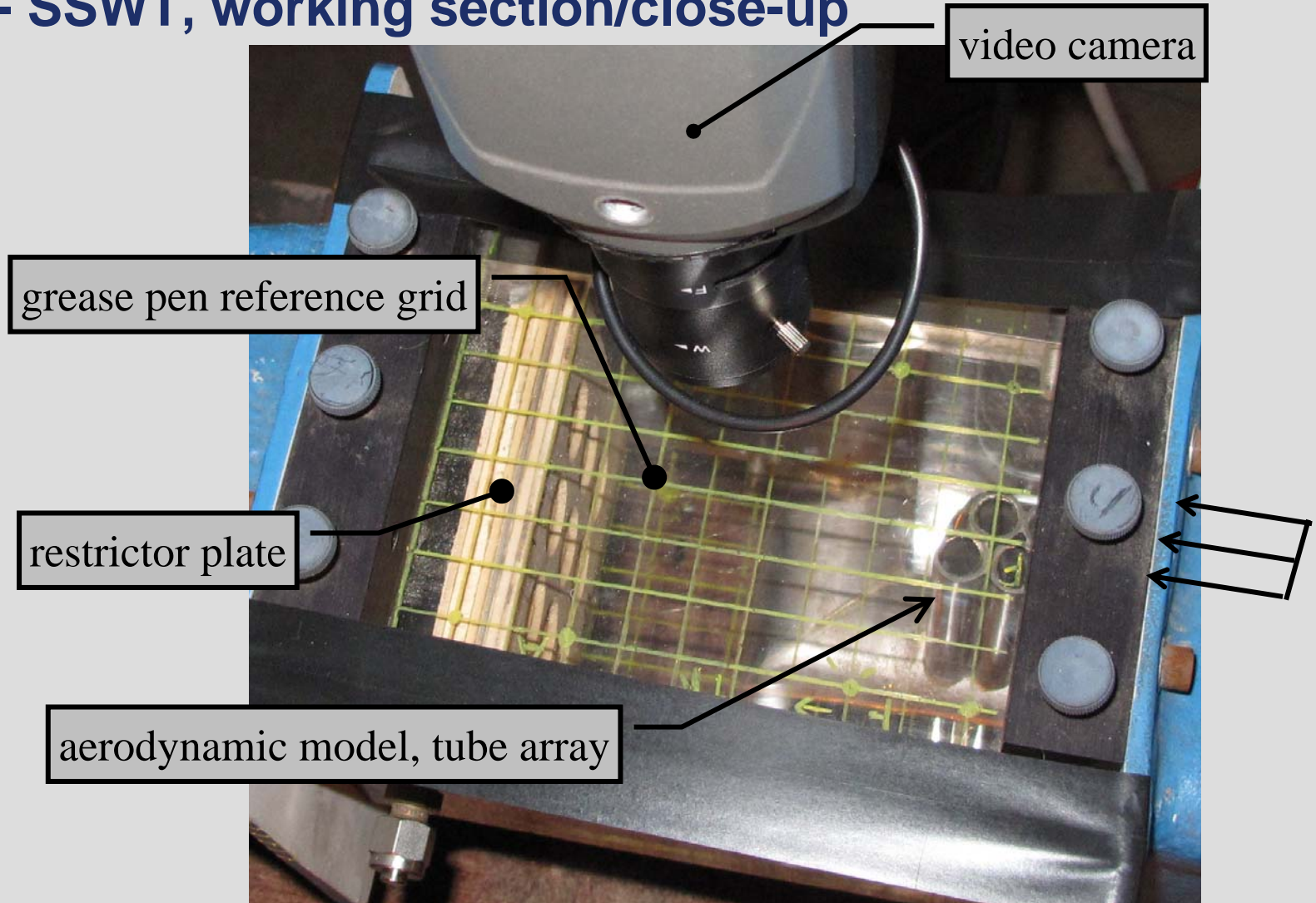
Flow visualization, SSWT Activity

Imagery - SSWT, tube array model



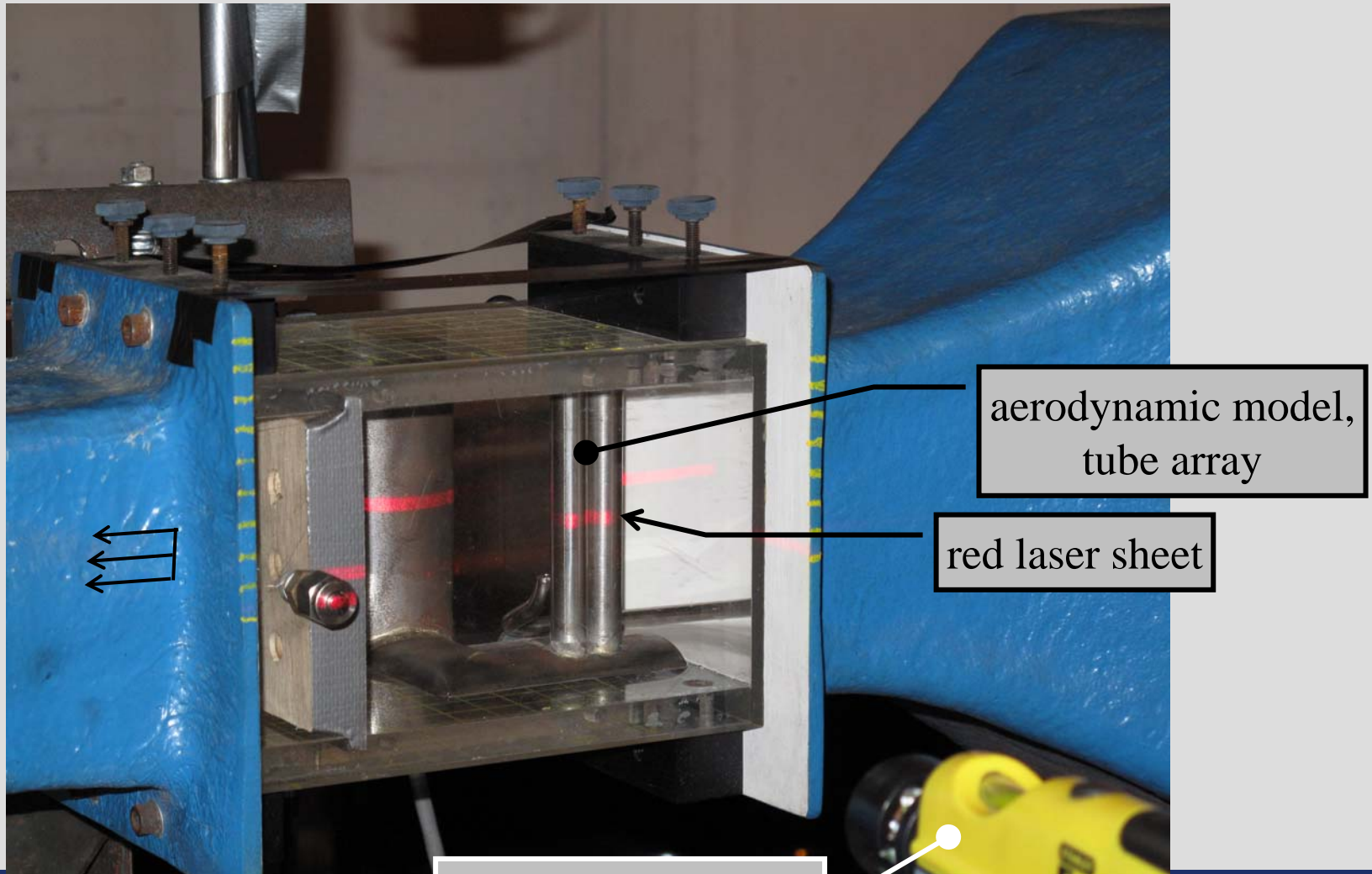
Flow visualization, SSWT Activity

Imagery - SSWT, working section/close-up



Flow visualization, SSWT Activity

Imagery - SSWT, working section/close-up



Flow visualization, SSWT Activity

Imagery - SSWT, set for work



Flow visualization, SSWT Activity

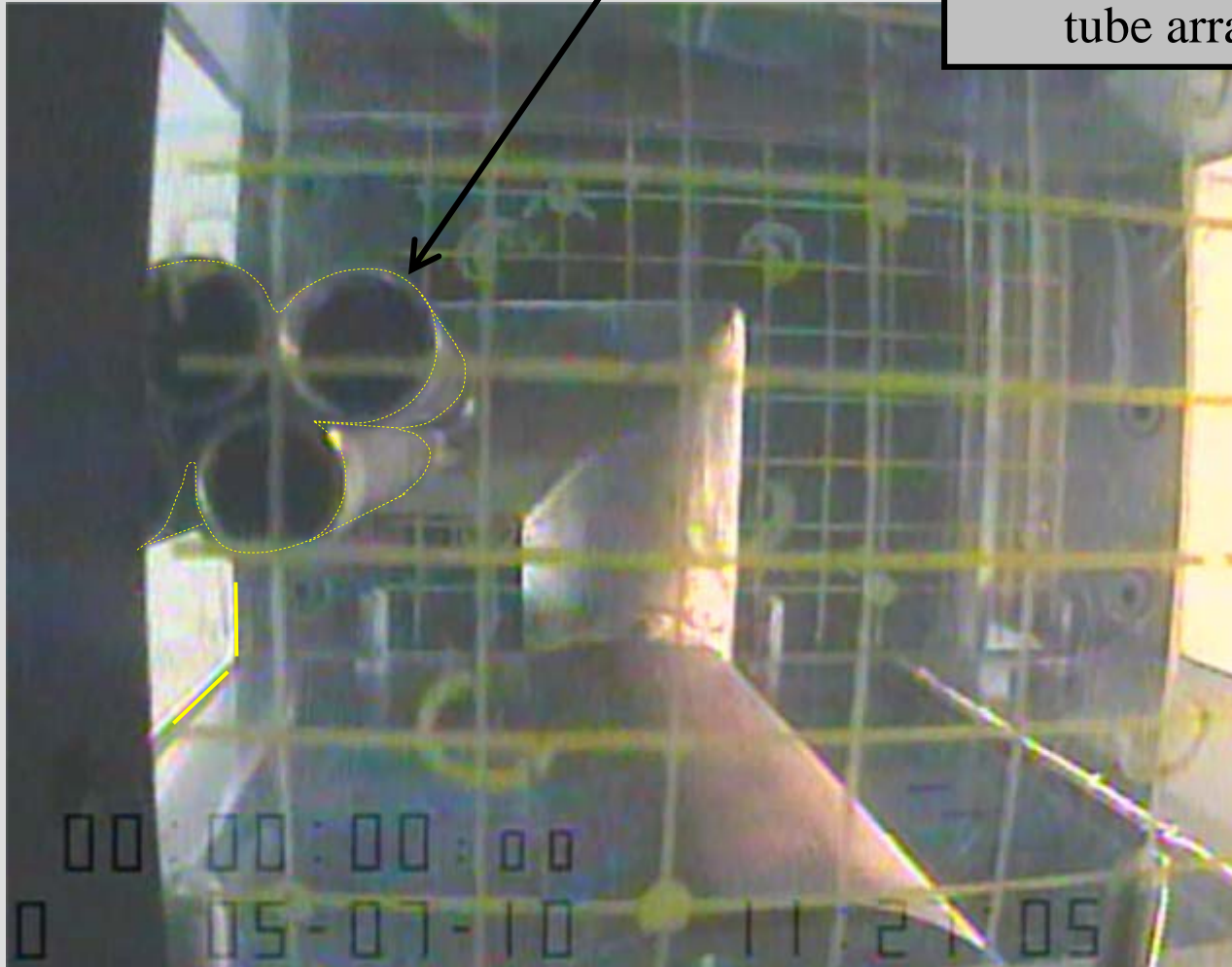
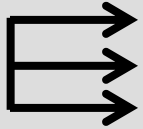
Tests 2009506-18 & -19, smoke visualizations

- **Common characteristics**
 - Air temperature = ~ 64°F (18° C)
 - Calculated air speed = ~ 47 ft/sec (14.3 m/s)
 - Aerodynamic model = tube array
- **Test 2009506-18 (outside wake region)**
 - Smoke delivered through the SSWT inlet forward of model
 - Smoke traversed from right to left (viewed here as top to bottom)
- **Test 2009506-19 (inside wake region)**
 - Smoke delivered to the wake of the model
 - 3 horizontal planes viewed

Flow visualization, SSWT Activity

Imagery - SSWT, orientation

RIGHT
FWD



aerodynamic model,
tube array

Flow visualization, SSWT Activity

Imagery - SSWT, smoke/inlet delivery, "clear" air/right side



Flow visualization, SSWT Activity

Imagery - SSWT, smoke/inlet delivery, right model boundary



Flow visualization, SSWT Activity

Imagery - SSWT, smoke/inlet delivery, centered



Flow visualization, SSWT Activity

Imagery - SSWT, smoke/inlet delivery, left model boundary



Flow visualization, SSWT Activity

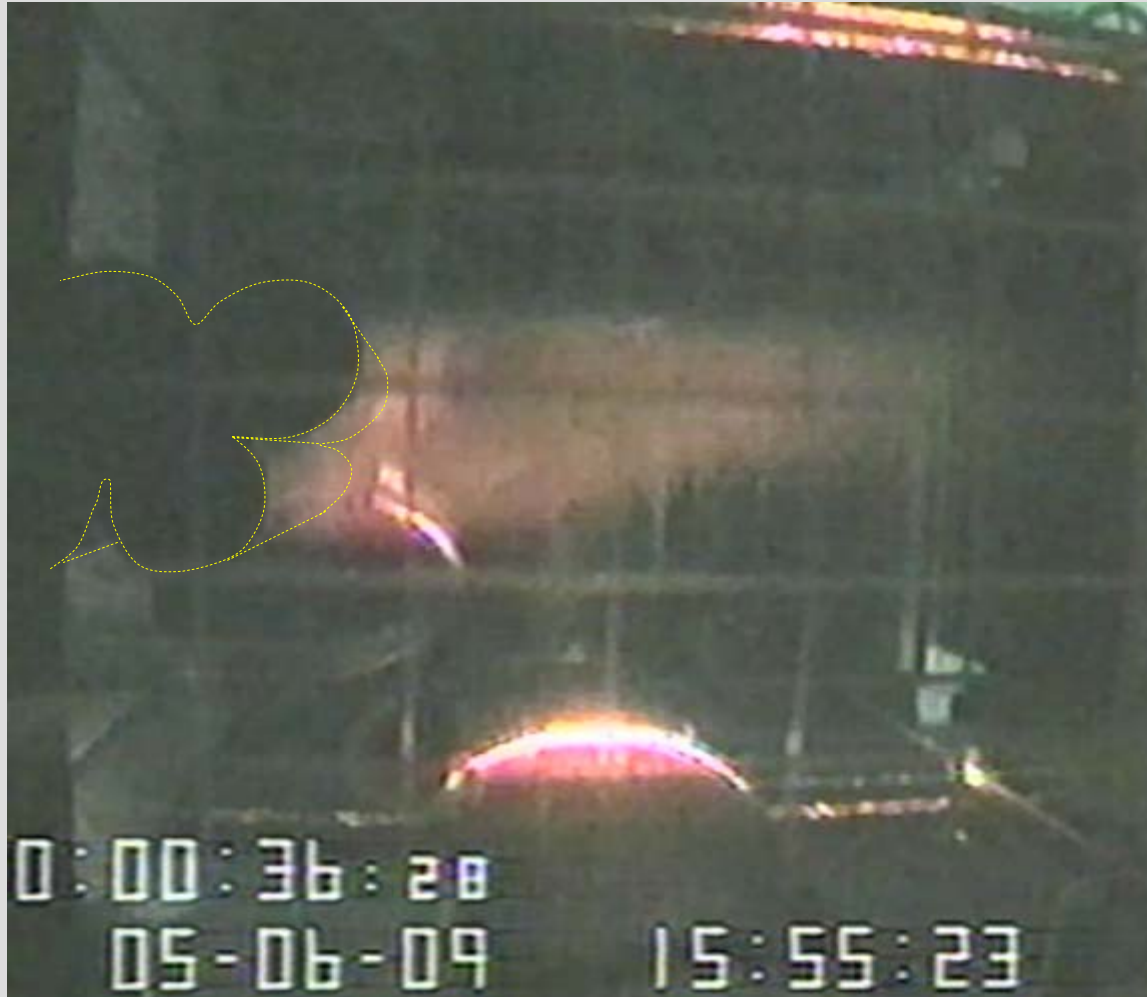
Imagery - SSWT, smoke/inlet delivery, "clear" air/left side

laser sheet is shadowed by the model



Flow visualization, SSWT Activity

Imagery - SSWT, smoke/wake delivery



Flow visualization, SSWT Activity

Imagery - SSWT, smoke/wake delivery



Flow visualization, SSWT Activity

Imagery - SSWT, smoke/wake smoke delivery



End

- **Acronyms**

MPSe = Minimum Performance Standard for Civil Aircraft Engine Nacelle & APU Compartments

APU = Auxiliary Power Unit

SSWT = small-scale wind tunnel

NFS = nacelle fire simulator, located at the FAA WJ Hughes Technical Center

rev = revision

fwd = forward

